

Suppressing ripple artifact while preserving accuracy in the
computation of acoustic fields with the angular spectrum method
using spatial filters

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1 **Abstract:** The angular spectrum method (ASM) is an effective tool
2 for propagating wavefields between parallel planes through decomposi-
3 tion of the field into a series of independent plane waves. One source of
4 error is interference from mirror sources introduced through the inher-
5 ent periodicity of the fast Fourier Transform (FFT) used to implement
6 this method numerically. Here, spatial filters were applied to atten-
7 uate waves propagating at large angles which are sensitive to mirror
8 sources. Simulations show that this suppresses the ripple artifact while
9 preserving the accuracy of the ASM-computed fields. To achieve com-
10 parable performance without filtering requires a 13.5-fold increase in
11 computation time.

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1. Introduction

Angular spectrum methods (ASM) are widely used in acoustics and other areas to propagate wave fields. Based on Fourier acoustics, the technique can be viewed as the decomposition of the wave field into a series of plane waves that are propagated independently and then superposed to obtain the field at the required axial depth. The approach can be utilised to characterise the pressure fields from a range of different source configurations including planar¹ and spherical² single-element transducers as well as multi-element arrays^{3,4}. It has also been extended to model wave propagation in layered homogeneous^{1,5} and inhomogeneous⁶ media. Propagation can be carried out in both the forward and backward directions where the former has applications to transcranial ultrasound focusing^{7,8} and thermal ablation treatment planning^{3,9} and the latter, so-called back-propagation, can be used to predict surface vibrations of acoustic sources often referred to as acoustic holography¹⁰.

The ASM approach is computationally advantageous over Rayleigh-Sommerfeld integral methods computed in the spatial domain as the convolution becomes a multiplication in the frequency domain which is computationally efficient through the use of the FFT¹¹. However, the FFT results in mirror sources caused by its inherent periodicity which can distort the calculated field and result in ripple artifacts appearing in the resulting field. These errors can be minimised by the careful selection of the spatial discretisation and domain size which in turn restricts the angular range. This has been previously discussed extensively¹²⁻¹⁵ where too large an angular range, resulting from increasing the number of grid points, or decreasing the domain size, results in a Fourier transformed Green's function that is un-

33 dersampled and hence introduces ripples into the solution. Conversely, limiting the angular
 34 range and thus truncating the homogeneous wave components to prevent undersampling
 35 will increase the accuracy in the far field at the expense of the near field by removing the
 36 evanescent components^{14–16}. Moreover, rather than controlling the angular range through
 37 the choice of sampling parameters, the removal of higher frequency plane wave components,
 38 termed angular restriction, can also increase accuracy¹⁷.

39 Here, we explore the use of different spatial filters to attenuate plane wave components
 40 propagating above a particular depth dependent critical angle. Two different filters: a top-
 41 hat filter and a Hamming window (HW), are compared and the mean squared errors relative
 42 to analytical or numerical solutions between filtering methods are compared.

43 2. Methods

44 2.1 Angular spectrum implementation

45 We assume a time-harmonic velocity potential of the form $\tilde{\phi} = \phi(x, y, z)e^{-i2\pi ft}$, which results
 46 in the Helmholtz equation in terms of the spatial velocity potential, ϕ :

$$\nabla^2\phi + k^2\phi = 0, \tag{1}$$

47 where $k = 2\pi f/c_0$. Taking 2D spatial Fourier Transforms (FTs) and assuming propagation
 48 in the $+z$ direction, the solution for the field at any other parallel plane is given by:

$$\Phi(u, v, z) = G(u, v)e^{ik_z z}, \tag{2}$$

49 where u, v are the spatial frequencies in the x, y directions, $G(u, v)$ is the FT of the xy source
 50 plane (i.e. the Angular Spectrum of the source) and $e^{ik_z z}$ is the propagation factor which
 51 independently propagates each plane wave through a distance, z , based on the projection
 52 of the wavenumber in the z -axis, $k_z = \sqrt{k^2 - (2\pi)^2(u^2 + v^2)}$. Source conditions can either
 53 be entered as pressure or normal velocity distributions, which are then converted into the
 54 velocity potential, ϕ through $p = -i2\pi f\phi$ and $u = ik_z\phi$.

55 2.2 Filtering

56 The geometry of the domain is shown Fig. 1. The angle of each plane wave, θ , with respect
 57 to the axial coordinate, z , following plane wave decomposition can be calculated as follows,

$$\theta = \arccos\left(\frac{k_z}{k}\right). \quad (3)$$

58 A critical angle, θ_c , can then be defined as the cutoff frequency for the filters based on the
 59 geometry of the domain, as shown in Fig. 1. Here, we have chosen the cutoff angle to
 60 correspond to a quarter of the domain width, w , from the centreline in each direction. As
 61 such, the critical angle decreases with increasing propagation depth and more components of
 62 the angular spectrum are filtered out at large propagation distances. For all simulations, no
 63 filtering was employed for $z < 2\lambda$ so as not to remove the exponentially-decaying evanescent
 64 components in the near field.

65 Two filtering methods were employed in order to suppress the interference from image
 66 sources: a top-hat function and a HW of order 192 as shown in Fig. 1B. The top-hat function
 67 (T) was implemented as shown in Eq. (4)

$$T(\theta) = \begin{cases} 1, & \text{if } \theta < \theta_c, \\ 0, & \text{if } \theta \geq \theta_c. \end{cases} \quad (4)$$

$$\theta_c = \arctan\left(\frac{0.25 * w}{z}\right), \quad (5)$$

68 and the HW was generated using the MATLAB function `fir1`.

69 2.3 Error estimation

70 The accuracy of the ASM simulations was compared to expressions from the Rayleigh integral
 71 assuming a time-harmonic source of the form $u = u_0 e^{j\omega t}$. The on-axis solution is given by¹⁸

$$p_r = \rho_0 c_0 u_0 [e^{-ikz} - e^{-ik\sqrt{z^2+a^2}}]. \quad (6)$$

72 For radial distributions, the Hutchins solution was used which requires numerical evaluation
 73 of a line integral¹⁹.

74 The mean squared error, MSE, was calculated between the simulated ASM pressure
 75 (\bar{p}) and the expected pressure value (p), summed and then averaged over the total number

76 of grid points, N , in either the axial direction (up to the Rayleigh distance) or in the radial
 77 direction (out to the second null in the field in the transverse plane at the Rayleigh distance):

$$MSE = \frac{1}{N} \sum_N (\bar{p} - p)^2. \quad (7)$$

78 The error function was not normalised.

79 *2.4 Simulations*

80 ASM simulations were computed for two different piston source conditions with uniform
 81 normal velocities:

82 1. 1 MHz, $a = 10\lambda$.

83 2. 1 MHz, $a = 2.5\lambda$.

84 For each source condition, simulations were conducted with a fixed grid size of 1024^2 and
 85 domain width (w) of either $4a$, $10a$ or $20a$. Wavefronts were propagated in steps of $z_{rayl}/1000$
 86 up to z_{rayl} (the Rayleigh distance) with either no filtering, a top-hat filter or a HW filter as
 87 defined in Section 2.2. Further simulations without filtering were also carried out with the
 88 same grid spacing but with increased zero-padding corresponding to domain widths of $40a$,
 89 $80a$ and $160a$.

90 **3. Results**

91 Fig. 2A–C displays typical pressure distributions for an unfocused, circular piston source
 92 vibrating with a uniform normal velocity. Diffraction from the source is evident in the xz -
 93 plane (Fig. 2A) and manifests as a series of rings in the two radial planes (Fig. 2B–C).

94 Consequently, for planes further into the farfield it is apparent that more energy will leak
 95 into the solution from mirror sources. Below each of the planar images the corresponding line
 96 scans are depicted comparing ASM with and without filtering to the analytical expression.
 97 It can be seen that for the unfiltered ASM simulation, ripples creep into the on-axis field
 98 at approximately a third of the Rayleigh distance and into the radial field at approximately
 99 one source radius out from the centreline, whereas the filtered method agrees well with the
 100 analytical expression throughout.

101 Fig. 3A–C show the MSE for the $a = 10\lambda$ source for a fixed number of grid points
 102 whilst varying the domain width for the unfiltered, top-hat filtered and HW filtered ASM
 103 simulations. For domain widths of $4a$, the spacing between mirror sources is insufficient to
 104 prevent large errors being introduced into the field even with filtering. However, at larger
 105 domain sizes, filtering proves an effective tool at reducing the errors with the HW filter
 106 performing better than the top-hat filter reducing the mean squared error by 4-fold for the
 107 axial, 2.2-fold for the radial slice at z_1 and 1.02-fold at the second radial slice z_2 for the $20a$
 108 domain width. The addition of the filters does introduce an additional computational cost
 109 of 1.7 times and 2.5 times for the top-hat and HW filters, respectively, as shown in Fig. 3D.

110 In general, it is possible to improve the accuracy of the unfiltered ASM by increasing
 111 the domain size and hence the spacing of the mirror sources. The question arises then, of
 112 how large a domain size is necessary to match the performance of the filtered ASM results?
 113 To that end, unfiltered ASM simulations with constant grid spacing but increasing domain
 114 width were computed, until the errors were comparable to the HW filtered results for a $20a$

115 width. It was determined that a width of $80a$ was required to achieve a similar level of
 116 accuracy for the on-axis distributions (Fig. 3E). However, in the two radial planes the errors
 117 in the unfiltered results were still 2.6-fold and 1.6-fold greater, respectively, (Fig. 3F–G) in
 118 spite of the 4-fold increase in mirror source spacing. Also, the unfiltered $80a$ case took 2.3
 119 times longer to calculate than the HW filtered $20a$ computation (Fig. 3H).

120 An interesting anomaly was encountered when increasing the domain size to $160a$ for
 121 the unfiltered simulation. In this case, the errors increased substantially relative to the $80a$
 122 case particularly for the on-axis distribution. The cause of this anomaly was found to rest
 123 with the conversion of the velocity source condition into the velocity potential through spatial
 124 differentiation, or division by ik_z , in the frequency domain. Consequently, at the evanescent
 125 wave boundary the magnitude of the transverse wavenumber equals the wavenumber itself,
 126 and thus $k_z = 0$ and a singularity occurs. In most cases, the sampled wavenumber space
 127 does not directly coincide with the evanescent boundary and so a singularity is avoided but
 128 there is still amplification of nearby wavenumber components which introduces the observed
 129 errors. This problem can be averted by implementing an additional band-reject filter or
 130 mask to remove components close to the evanescent wave boundary. It is also noted that
 131 these problems do not occur for a pressure-source where division by k_z is not required.

132 Similar trends were observed for a smaller source radius, $a = 2.5\lambda$ as shown in Fig.
 133 4 which shows the data at z_1 in which the error was 1.7 times lower for the HW filter
 134 than the top-hat filter. When increasing the domain size for the unfiltered simulations, the
 135 anomalous error rise for $160a$ seen in Fig. 3E–G was not observed and this domain width

136 offered comparable radial errors to the $10a$ HW filter but at an increased computational
137 expense of 13.5-fold (Fig. 4D).

138 **4. Conclusion**

139 This paper has described the use of spatial filters with the ASM to suppress the ripple artifact
140 from mirror sources by restricting the angular range of the ASM in a depth dependent
141 manner, to remove components that propagate at large angles. Two filters were considered, a
142 top-hat filter, and a Hamming window filter, with the latter providing improved performance.
143 To match the performance achieved using a Hamming window filter with an unfiltered ASM
144 approach, domain widths of up to 8 times larger are required resulting in an increased
145 computational cost of between 2.3 and 13.5-fold. Therefore, this motivates the use of spatial
146 filters as an effective method to improve the accuracy of ASM simulations whilst reducing
147 computational times.

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208 **Figure Captions**

209 **Figure 1:** Schematic of ultrasound propagation from circular piston source (shown in grey).

210 (A) Transverse (xy) plane of width, w . (B) Axial (yz) plane from source to propagation
 211 planes, $z_1 = 0.334z_{rayl}$ and $z_2 = z_{rayl}$. Filter cutout frequency is defined based on the angle
 212 of the wavevector, θ , with two different low-pass filters: a top-hat function (black) and a
 213 Hamming window (grey).

214 **Figure 2:** Typical pressure distributions relative to the source pressure, $p_0 = \rho_0 c_0 u_0$, for an
 215 unfocused piston source with and without Hamming window (HW) filtering. Filtered fields:
 216 (A) xz -plane, (B) xy -plane at $z_1 = 0.334z_{rayl}$, (C) xy -plane at $z_2 = z_{rayl}$. (D) Axial field
 217 comparison between on-axis Rayleigh solution, Eq. (6), and ASM method with and without
 218 HW filtering. (E,F) Radial field comparison between Hutchins solution and ASM method
 219 with and without HW filtering at z_1, z_2 (log-scale).

220 **Figure 3:** Mean squared errors for a circular piston source of radius, $a = 10\lambda$ for unfiltered
 221 (N), top-hat (T) filtered and Hamming window (HW) filtered ASM simulations. Top row:
 222 fixed number in grid points for each plane, 1024^2 , for three domain widths, $4a, 10a$ and $20a$.
 223 Bottom row: comparison of $20a$ domain width simulations with larger unfiltered simulations
 224 of $40a, 80a$ and $160a$. (A, E) On-axis field errors. (B, F) Radial field errors at $z_1 = 0.334z_{rayl}$.
 225 (C, G) Radial field errors at $z_2 = z_{rayl}$. (D, H) Computational times normalised to HW $20a$
 226 case. The reason for the anomalous result for $160a$ in E–G is discussed in the text.

227 **Figure 4:** Mean squared errors for a circular piston source of radius, $a = 2.5\lambda$ for unfiltered
 228 (N), top-hat (T) filtered and Hamming window (HW) filtered ASM simulations. Top row:
 229 fixed number in grid points for each plane, 1024^2 , for three domain widths, $4a, 10a$ and $20a$.
 230 Bottom row: comparison of $20a$ domain width simulations with larger unfiltered simulations
 231 of $40a, 80a$ and $160a$. (A, C) Radial field errors at $z_1 = 0.334z_{rayl}$. (B, D) Computational
 232 times normalised to HW $20a$ case.