

# Modified Generalized Maxwell Model for Hysteresis Behavior of Elastomeric Dampers

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## ABSTRACT

A constitutive model is presented for the hysteretic behavior of elastomeric material, called the Modified Generalized Maxwell Model (MGMM). A new force-displacement relationship which describes the well known Generalized Maxwell Model (GMM) is also proposed in this study, and forms the basis for the MGMM. This relationship can be used regardless the number of Maxwell elements. The damper was first characterised using terms of Visco-Elastic (VE) material, determining its shear storage modulus and loss factor for a range of cyclic tests. This led not only to an understanding of how the main mechanical properties of the damper change when strain amplitude, frequency, and temperature change, but also to a more detailed understanding of how the hysteresis shape of the material change along the alteration of these parameters. The final model was calibrated using test data obtained from sweep amplitude tests over a range of frequencies, and ambient temperatures and was able to accurately predict the dynamic performance of the dampers.

## INTRODUCTION

Passive energy dissipation devices have the ability to improve the seismic behavior of buildings (Makris and Constantinou 1991; Shen and Soong 1995; Soong and Dargush 1997; Fan 1998; Lee 2003; Vargas and Bruneau 2007; Shrimali et al. 2015) by reducing drift, force and deformation

24 demands on the structural elements, which are responsible for providing lateral load resistance, in  
25 addition to reducing velocity and acceleration demands on non-structural components (Karavasilis  
26 et al. 2012). This paper focuses on the evaluation of the hysteresis behavior of two Elastomeric  
27 Dampers (EDs), developed by TARRC (the Tun Abdul Razak Research Center, UK).

28 Elastomers, like many other polymers, show viscoelastic properties, and hence highly depend  
29 on the strain amplitude, frequency, and ambient temperature. However, (Lee 2003) showed that  
30 elastomeric dampers are less sensitive to frequency compared to conventional viscoelastic materials,  
31 a fact which makes them even more effective. Moreover, a distinct feature of the hysteresis loops  
32 of elastomers is that softening at a small strain amplitude and hardening at a relatively large strain  
33 amplitude may occur in the amplitude range of practical interest (Sause et al. 2007). Similar  
34 softening-hardening hysteresis behavior was observed for elastomeric dampers (Sause et al. 2007),  
35 and for high damping rubber base isolators (Taniwangsa and Kelly 1996; Aiken 1997). Another  
36 interesting characteristic of the elastomeric materials, known as the Mullin's effect, is the softening  
37 of the material under the first few cycles of deformation, after which the stress-strain behavior  
38 approaches a steady state (Hepburn and Reynolds 1979). The main properties of the elastomers  
39 can be summarized as: exhibit VE behavior especially under relatively small strains, extremely  
40 high extensibility, complete recovery after removal of the imposed stress, and strain hardening  
41 at relative large strains. These are advantageous for passive dissipation devices since, even after  
42 strong earthquakes, the elastomeric dampers will not exhibit any damage or permanent deformation,  
43 resulting in more economical solutions.

44 Most elastomer applications for seismic protection of structures have been implemented for  
45 seismic isolated structures (Koh and Kelly 1990; Papoulia and Kelly 1994; Shrimali et al. 2015;  
46 Silwal et al. 2015) in the form of elastomeric bearings for base isolation. There are a few applications  
47 of VE dampers which were used retrofitting structures connecting the damper with the braces. The  
48 main target of the first implementations of the VE dampers were to reduce the wind induced  
49 vibrations. Therefore, one of the first applications of VE dampers was in the World Trade Center  
50 (1969) in order to reduce the acceleration levels due to wind (Mahmoodi et al. 1987). Hence, about

51 10000 VE dampers were installed in each tower from the 10th to the 110th floor. Their design was  
52 carried out in such a way to assist the steel frame mitigating the wind induced movement. They were  
53 located between the lower chords of the horizontal trusses and the the outer columns of the structure.  
54 The total achieved damping was calculated and found to be in the range of 2.5%-3% of critical.  
55 Seismic applications started a lot later, and more specifically in 1993 where the 13 story Santa  
56 County building in San Jose, CA, was decided to be retrofitted, since it was found that the viscous  
57 damping in the fundamental mode was less than 1% of critical. VE dampers were chosen as final  
58 solution, since they could provide increased damping for both frequent low-level ground shaking  
59 and strong ground motions. Two VE dampers were added to each building face per floor, increasing  
60 the fundamental damping to 17% of the critical. More recent applications of VE dampers connected  
61 with steel diagonal braces include the Beijing 7 Star Morgan Plaza Hotel C (China, Beijing) in 2007  
62 where 108 viscous and viscoelastic dampers were added for both wind and earthquake protection,  
63 and the Hotel Stockton (USA, Stockton, CA), where a combination of both viscous and viscoelastic  
64 dampers were added to reduce the structure's response due to seismic loads (Kit Miyamoto and  
65 Lon M. Determan ). Another different approach was followed by Rant D.R. (Pant et al. 2017),  
66 where a 110-story, 630 m mega-tall building was retrofitted with VE dampers, which replaced the  
67 coupling beams connecting the shear walls. The main concept behind this decision is that since the  
68 deflected shape of very tall buildings due to horizontal loads is similar to cantilever, rather than the  
69 shear deflection of moment resisting frames which consists the main resisting system of low-rise  
70 buildings. Hence, the whole dampers philosophy was adjusted as well. The dampers replaced  
71 about 60% of the diagonally reinforced concrete coupling beams of the original structure. The  
72 results showed that the peak inter-story drift ratios were reduced by up to 15% under the Maximum  
73 Considered Earthquake, and the peak floor accelerations by approximately 24%. One of the rare  
74 applications of rubber-like dampers connected with the surrounding structure with diagonal braces  
75 was carried out by Teramoto T. (Teramoto et al. 1996) on a 11 story, 44 m tall building. The main  
76 target of the added dampers was to reduce the vibrations due to both seismic loads and traffic motion  
77 since the building was close to subway lines. The results improved the building's performance

78 by reducing both the maximum accelerations and story drifts by approximately 20%. Another  
79 seismic retrofit was carried out in the Gentile-Fermi School in Fabriano, after the building suffered  
80 sufficient damage during an earthquake in the Umbria-Marche region (September 1997). A total  
81 of 33 elastomeric dampers were added at the second and the third floor of the building absorbing  
82 50% of the input energy.

83 Numerous models have been proposed in order to simulate the hysteresis behavior of elastomeric  
84 materials as passive dissipative devices. Lee (2003) proposed rate-independent and rate-dependent  
85 models based on asymptotic functions, simulating softening and hardening behavior which the  
86 elastomer experiences, using second and third order polynomials. A linear dashpot was also  
87 added to the proposed model in order to take into account rate dependent effects. However, the  
88 results were not satisfactory for the entire range of frequencies which were used, while also some  
89 discrepancies were noticed between the experimental and analytical values regarding the equivalent  
90 shear modulus and especially the loss factor. Kästner et al. (2012) proposed a linear combination  
91 of rate independent and rate dependent components as a way of modelling the inelastic material  
92 behavior. Even though good results were achieved for relaxation and tensile tests, the comparison  
93 with cyclic loading showed that the model overestimates the hysteresis. Shen and Soong (1995)  
94 used Boltzmann's superposition principle (Ferry 1980). The model was only valid under the  
95 assumption of linearity, and limitations of frequency below 3Hz, and shear strain amplitude below  
96 20%, where viscoelastic effects are dominant. Generalized Maxwell (or Kelvin) models (GMM)  
97 (McCrum et al. 1997; Karavasilis et al. 2011) and Fractional-Derivative models (Koh and Kelly  
98 1990; Makris and Constantinou 1991) have been also proposed to capture rate dependent effects  
99 of passive dissipative dampers. The Bouc-Wen model has been also used (Constantinou and  
100 Tadjbakhsh 1985; Yu et al. 2016), especially in the case of elastomeric materials being used in  
101 base isolation. Petrone et al. (2004) proposed an assembly of springs, dashpots, and Coulomb  
102 friction sliders in order to represent the dynamic characteristics of elastomers in both in series  
103 and in parallel configuration. Comparison was made based on stiffness-frequency and damping-  
104 frequency terms. Jrad et al. (2013) used a non linear GMM for viscoelastic representation with

105 satisfying accuracy in frequency domain. He has also proposed a model which couples GMM with  
106 a friction model developed by Dahl (Dahl 1968), with promising results (Jrad et al. 2017) in an  
107 attempt to simulate elastomeric rotational joint behavior under harmonic loading. However, the  
108 conventional way of modeling elastomeric behavior, at least for design purposes, is the simpler one  
109 that FEMA (Prestandard 2000) proposes, where an equivalent stiffness and loss factor are assumed  
110 for the elastomer based on the natural frequency of the building. Moreover, Silwal et al. (2015) used  
111 a ViscousDamper material based on the Maxwell material property, in order to model elastomeric  
112 behavior as part of a hybrid passive damper.

113 To determine the main characteristics of elastomeric dampers as passive dissipation devices,  
114 shear storage modulus and shear loss factor (McCrum et al. 1997; Soong and Dargush 1997) should  
115 be obtained for a range of strain amplitude, rate of loading, and ambient temperature. These values  
116 usually form the basis to design a structure with dissipative systems in order to achieve an advanced  
117 performance level, which will lead the structure not only to be able to withstand strong earthquakes  
118 without severe damage, but also to maintain its functionality. In contrast, the common design  
119 approach prevents the structure from collapse, but at the same time energy dissipation is expected  
120 to occur in specially designed ductile plastic hinge regions of beams and columns bases. This design  
121 approach has been considered acceptable because of economic considerations provided, of course,  
122 that structural collapse is prevented and life safety is ensured (Constantinou et al. 1998). However,  
123 under large earthquakes the damage may not be restricted to the aforementioned regions, but can  
124 extend to the majority of the columns leading to permanent deformation in undesired locations with  
125 potential catastrophic and uneconomical results. Hence, structures are often desired to achieve a  
126 more advanced performance level even due to strong earthquakes, especially if these structures are  
127 defined as highly important, such as hospitals, schools, and museums.

128 This paper examines the effect of elastomeric materials when used as discrete dampers in the  
129 form of a layer bonded between two steel plates. The design philosophy is to implement the dampers  
130 throughout the height of the structure, with dampers inserted between the connection of additional  
131 diagonal braces and the middle of the beam of the upper floor. The dissipation mechanism is

132 triggered through shear deformation of the material, as the building moves due to dynamic loading.

133 A differential time-domain equation to accurately simulate the shear force-displacement be-  
134 havior of the elastomer, based on the Generalized Maxwell Model (GMM), is presented here.  
135 Apart from the traditional characterization tests, sweep amplitude displacement tests at different  
136 frequencies were also carried out in order to form the basis of the data used to calibrate the model.  
137 Furthermore, an extension to the model is proposed in order to capture dynamic behavior of the  
138 damper under different ambient temperatures. Part of the proposed hysteresis model is a new  
139 equation which describes the force-displacement relationship of the GMM in the time domain for  
140 any given displacement and any given value,  $N$ , of the Maxwell elements which is updated for  
141 every time-step.

## 142 **ELASTOMERIC DAMPER AND CHARACTERIZATION TESTS**

143 Two elastomeric dampers were provided by the Tun Abdul Razak Research Centre (TARRC).  
144 They consist of a rubbery material, consisting of natural rubber, carbon black, and resin, bonded  
145 between steel plates. The cure for material test-pieces was 30 minutes at 150°C. Material properties  
146 can be seen in Table 1.

147 Each of these dampers has overall dimensions of 260x260 mm<sup>2</sup>, of which only 180x230 mm<sup>2</sup>  
148 corresponds to the elastomer's dimensions while the materials' thickness is 11.75 mm (fig.2). The  
149 230 mm dimension coincides with the loading direction. For testing, each elastomeric damper was  
150 connected to additional steel plates,  $A$  (fig.1), which were connected to a stable reacting wall. The  
151 interior plate of the damper was connected to a plate,  $B$  (fig.1), which in turn enables connection  
152 to the actuator via a central plate,  $C$  (fig.1). The final configuration of the rig can be seen at Fig.1.

153 This arrangement allowed the two dampers to be tested simultaneously in shear in a symmetric  
154 arrangement. Loads were applied by a 100 kN servo-hydraulic actuator. A thermo-couple was also  
155 attached to the elastomer in order to measure the ambient temperature, and to capture any increase  
156 in temperature during the tests. It is well known that rubber-like materials exhibit non linear  
157 behavior that depends on strain amplitude, ambient temperature, loading frequency, and loading  
158 history (Lee 2003). Hence, the experimental tests were focused on verifying non linear dependence

159 on these parameters. Among these parameters, it is has been shown that strain dependence is the  
160 most dominant factor (Papoulia and Kelly 1994; Lee 2003).

161 In order to investigate the effect of these parameters a series of tests was carried out based  
162 on sinusoidal displacement histories at the EDs. In order to avoid large initial displacement and  
163 velocity, ramping cycles were implemented, along with 18 full sinusoidal cycles. This process was  
164 repeated for frequencies 0.25, 0.5, 1.0, 2.0, 3.0, and 4.0Hz, for strain amplitudes 10, 20, 30, 40,  
165 and 50%, and for ambient temperatures 20, 25, 30, and 35°C. The maximum shear strain amplitude  
166 was kept within 50%, to avoid any potential permanent damage, severe cracking, or debonding of  
167 the elastomer.

Elevated temperature tests were achieved with the use of a heater, and a temperature control chamber, so that the target temperature was kept stable during the test. During testing it was noticed that after the first loading cycles, the hysteresis loops became more and more steady and approximately repeatable, corresponding to smaller and smaller stiffness degradation. Hence, in order to determine the mechanical properties of the dampers the average value of the hysteresis loops of cycles 5-15 was taken. The shear storage modulus,  $G'$ , and either the shear loss modulus,  $G''$ , or the loss factor,  $n$ , govern the main characteristics of the dynamic behavior of viscoelastic and elastomeric materials. All the above parameters can be extracted from each of the hysteresis loops as:

$$G' = \frac{\tau(\gamma=\gamma_{max})-\tau(\gamma=\gamma_{min})}{\gamma_{max}-\gamma_{min}} \quad (1)$$

$$n = \frac{1}{2\pi} \frac{ED}{ES} \quad (2)$$

where  $\tau(\gamma = \gamma_{max})$  is the shear stress corresponding to the maximum strain  $\gamma_{max}$ , and  $\tau(\gamma = \gamma_{min})$  is the shear stress corresponding to the minimum strain  $\gamma_{min}$ .  $ED$  and  $ES$  are the dissipated energy per cycle of oscillation, and the maximum energy stored respectively.  $ED$  can be calculated based

on the area of the hysteresis loop and  $ES$  can be calculated as:

$$ES = \frac{1}{2}G'\gamma_{max}^2 \quad (3)$$

The values of both  $G'$  and  $n$  for each combination of frequency, amplitude and temperature are shown in Table 2. Equivalent viscous damping ratio can be also calculated as:

$$\zeta_{eq} = \frac{1}{4\pi} \frac{E_D}{ES} = \frac{n}{2} \quad (4)$$

And in terms of a Kelvin solid model,  $\zeta_{eq}$  can be also calculated as:

$$\zeta_{eq} = \frac{G'\omega}{2G''} \quad (5)$$

168 The values of both the loss factor and the shear storage modulus are similar to those of rubbery  
169 materials used by (Lee 2003), and other researchers (Teramoto et al. 1996; Soong and Dargush  
170 1997).

171 As was anticipated, strain amplitude had a larger impact on the EDs' behavior than frequency.  
172 This can be more clearly demonstrated by comparing the amplitude dependence at a range of  
173 frequencies at fixed temperature, in the form of fig.3. This figure proves the above statement that  
174 the dominant factor which affects the elastomer's dynamic behavior is the amplitude and not the  
175 frequency, especially in the case of the loss factor, which practically remains the same regardless  
176 of any change in frequency. It is also noticeable that when the strain amplitude increases above  
177 30% (which is the key point, as will be shown later, for the transition between the viscoelastic and  
178 elastomeric phases) the rates of change of both the shear storage modulus and the loss factor reduce  
179 significantly. Below 30%, a large decrease, especially in the loss factor case can be observed.  
180 Furthermore, it seems that amplitude and frequency have exactly opposite effects, since increasing  
181 strain amplitude leads to decreasing the mechanical properties of the elastomer, while increasing  
182 the loading frequency leads to increasing them.

183 An alternative way of visualizing the elastomer's behavior is by plotting the hysteresis loops  
184 of the dampers at each shear strain amplitude, keeping the frequency and temperature constant.  
185 Hence, observing fig.4 it can be seen that the elastomer exhibits a behavior which can be rationally  
186 approximated as viscoelastic for strain amplitudes of 10%-30%, proving that elastomers exhibit  
187 viscoelastic behavior under relatively low amplitudes, which is one of the main material characteris-  
188 tics. The more the strain amplitude increases, the more the hysteresis behavior tends to form a more  
189 typical elastomeric shape (compare for example fig.4e, which has slightly different characteristics  
190 from the conventional viscoelastic shape, with fig.4a). On the other hand, this is not the case when  
191 the frequency is altered. In this case, even though the values of the mechanical properties change,  
192 the overall shape of the hysteresis loops remain the same.

193 When considering the elastomer's dependence on strain amplitude, the fact that not only the  
194 mechanical characteristics change, but also the fundamental behavior of the material, makes the  
195 modeling procedure of the material extremely difficult. But, at the same time it shows that when  
196 the material has larger strain amplitudes, and hence behaves more like a typical elastomer, it tends  
197 to be more and more independent of the frequency, in contrast with its viscoelastic phase when the  
198 strain amplitudes are relatively low.

199 With regard to the effect of ambient temperature, both the shear storage modulus, and the loss  
200 factor decrease with the increase of temperature. This can be schematically observed by plotting  
201 the hysteresis curves of the dampers for different temperatures, while keeping the frequency, and  
202 the strain amplitude constant. fig.5 shows how the dissipative capacity of the dampers becomes  
203 smaller as the ambient temperature increases. However, it can also be seen that the shape of  
204 the hysteresis remains unchanged, leading to the conclusion that only strain amplitude leads to  
205 different hysteresis shapes, in contrast with frequency and temperature, which affect, more or less,  
206 the dampers' characteristics but not their hysteresis shape.

207 The effect of the temperature on the elastomers' properties can be more clearly seen in fig. 6,  
208 where the values of loss factor and shear storage modulus are plotted against frequency for different  
209 temperatures, while keeping the strain amplitude constant at 40%. Two different conclusions can be

210 made through these figures: a) increase of temperature leads to decrease of dampers' effectiveness.  
211 However, the rate of this decrease decreases, as the temperature gets higher, b) the dampers'  
212 mechanical properties remain almost constant, especially in the case of the loss factor, with any  
213 frequency alteration.

214 Moreover, similar observations with the 20°C case, can be extracted from the experimental  
215 data with regard to the rate of decrease of both shear storage modulus and loss factor. For every  
216 given temperature,  $G'$  and  $n$  tend to alter in a similar way when frequency or strain amplitude  
217 changes, leading to the estimation that the elastomer behaves similarly in different temperatures.  
218 Even though its mechanical characteristics are reduced, at the same time it seems that any alteration  
219 in strain amplitude and frequency results in quite similar change in the loss factor, and the shear  
220 storage modulus.

## 221 **PROPOSED HYSTERESIS MODEL**

### 222 **Linear Generalized Maxwell Model**

223 The proposed model for the hysteresis behavior of the EDs is based on the well known Gen-  
224 eralized Maxwell Model (GMM). The GMM has been proposed as numerical representation of  
225 VE materials (McCrum et al. 1997; Karavasilis et al. 2011) and can be graphically represented  
226 in fig. 7 in force-displacements terms. It consists of linear spring, with stiffness  $k_0$ , in parallel  
227 with N Maxwell (spring-damper) elements. When stress-strain terms are being used instead, the  
228 parameters of fig. 7 have to be adjusted as well. In this paper force-displacements terms are used;  
229 however stress-strain terms have been used in literature (Lee 2003; Karavasilis et al. 2011) as well.

The most common procedure of determining the parameters of a VE or elastomeric material, when the GMM is used, is to minimize the error between the analytical values of the shear storage and the shear loss modulus, and the corresponding values which are based on characterization experiments. Based on the assumption that the applied strain has sinusoidal form, with a frequency  $\omega$ , the corresponding shear storage modulus,  $G'$  and the shear loss modulus,  $G''$  can be determined

as (Bland 1960):

$$G' = G_0 + \sum_{i=1}^N \frac{G_i \omega^2 \tau_i^2}{1 + \omega^2 \tau_i^2} \quad (6)$$

$$G'' = \sum_{i=1}^N \frac{G_i \omega \tau_i^2}{1 + \omega^2 \tau_i^2} \quad (7)$$

where

$$\tau_i = \frac{c_i}{k_i} \quad (8)$$

230 is known as the relaxation time.

231 The above procedure is based on minimizing the error, on frequency domain, between the  
232 analytical expressions of shear storage modulus and shear loss factor and their experimental values,  
233 for discrete loading frequencies and is valid only for sinusoidal loading. However, in order to be  
234 generalized and validated under random loading, as in the case of seismic loading, these values  
235 are normally determined based on a frequency range close to the frequency content of earthquakes  
236 (usually 0.25-4 Hz), which is also similar with the content of the natural frequencies of many  
237 structures.

238 Other researchers (Torvik and Bagley 1984; Koh and Kelly 1990; Shen and Soong 1995; Lai  
239 et al. 1996; Hwang and Wang 1998) who proposed different models than GMM, such as fractional  
240 derivatives model, used the same minimization technique in order to obtain the parameters of their  
241 models. However, lack of comparison of experimental data with simulation's models in a hysteresis  
242 form, (at least using the cyclic tests with the same frequencies and amplitudes under which these  
243 models were initially validated) does not allow the reader to have a better understanding about the  
244 behavior of the material under random dynamic loading and about the effectiveness of the proposed  
245 model. Shen and Soong (1995) in contrast, compared the hysteresis behavior of test and simulation  
246 data, where some discrepancies were noticed.

247 A different approach was used here, where the main focus of the GMM is on extracting the force-  
 248 displacement relationship in the time domain for any  $N$  number of Maxwell elements, and therefore  
 249 the model can be validated under any random loading, and potentially capture the hysteresis shape  
 250 of the elastomer. Although, it is very easy to extract this relationship when only one Maxwell  
 251 element is used, this becomes much more complicated when more elements are being used.

252 Initially, a force-displacement relation for the general case of  $N$  Maxwell elements was devel-  
 253 oped based on Laplace transformation. The full derivation is shown in Appendix A and only the  
 254 final result is presented here:

$$\begin{aligned}
 & F + \sum_{a=1}^N \left[ \frac{\partial^a F}{\partial t^a} \left\{ \sum_{m_i \in A_i} \left\{ \prod_{r \in B} \left( \frac{c_r}{k_r} \right) \right\} \right\} \right] = \\
 & \sum_{\beta=1}^N \left[ \frac{\partial^\beta u}{\partial t^\beta} \left\{ \sum_{x_c \in Q_c} \left\{ \prod_{z \in W} \left( \frac{c_z}{k_z} \right) \sum_{z \in W} k_z \right\} \right\} \right] + \left[ F_0 + \sum_{d=1}^N \left[ \frac{\partial^d F_0}{\partial t^d} \left\{ \sum_{p_v \in H_v} \left\{ \prod_{f \in L} \left( \frac{c_f}{k_f} \right) \right\} \right\} \right] \right]
 \end{aligned} \tag{9}$$

255 where:

- 256 •  $u$ ,  $F$ , and  $F_0$  represent the displacement, total force, and force corresponding to the linear  
 257 spring  $k_0$
- 258 •  $i \in \{1, 2, \dots, \alpha\}$ ,  $c \in \{1, 2, \dots, \beta\}$ ,  $v \in \{1, 2, \dots, d\}$
- 259 •  $A_i \in \{i, i+1, i+2, \dots, N-\alpha+i\}$ ,  $Q_c \in \{c, c+1, c+2, \dots, N-\beta+c\}$ ,  $H_v \in \{v, v+1, v+2, \dots, N-d+v\}$
- 260 •  $B \in \{m_1, m_2, \dots, m_{\alpha-1}, m_\alpha\}$ ,  $W \in \{x_1, x_2, \dots, x_{\beta-1}, x_\beta\}$ , and  $L \in \{p_1, p_2, \dots, p_{d-1}, p_d\}$
- 261 •  $m_i > m_{i-1}$ ,  $x_c > x_{c-1}$ ,  $p_v > p_{v-1}$  for  $i, c, v > 1$

262 The same process is often carried out with stress-strain, instead of force-displacement terms,  
 263 although here the latter is preferred. This equation, which is actually a combination of force and  
 264 displacement derivatives, is valid for every displacement input and for every value of  $N$  elements,  
 265 regardless of the frequency, which represents its main advantage over the classical method described  
 266 earlier.

267 The advantage of Eq. 9 compared with the traditional form of the GMM described earlier, apart

268 from the fact that it may be useful in the mathematical and physical fields, is that it can be used for  
269 random loading and not only sinusoidal loading, and the force-displacement relationship can now  
270 be extracted in the time domain for any number  $N$  of Maxwell elements, capturing the material's  
271 hysteresis shape, and is not based on minimizing the error between the analytical expressions of  
272 the shear storage modulus and their corresponding experimental values. Furthermore, it can be  
273 also concluded that Eq.9 is very similar with the force-displacement relationship of the Generalized  
274 Derivatives model (Nashif et al. 1985), which has also been used for modeling elastomeric behavior  
275 (Torvik and Bagley 1984; Koh and Kelly 1990; Hwang and Wang 1998). However, the unknown  
276 parameters of the Generalized Derivatives equation are now determined, and are functions of the  
277 stiffness  $k_i$ , and the damping coefficients  $c_i$  of the GMM. Replacing the integral derivatives with  
278 fractional derivatives, the generalised model is changed to the well known fractional derivative  
279 model (Nashif et al. 1985). Another restriction of either Generalized Derivatives or Fractional  
280 Derivatives model is the way they have been implemented in order to extract their parameters;  
281 the same minimization technique which was mentioned earlier for the GMM is used here as well  
282 (Nashif et al. 1985). Furthermore, even though Fractional Derivative models are usually based on  
283 fewer parameters, it was found that additional parameters were necessary to capture the dynamic  
284 characteristics of the EDs tested.

285 When more simplifying procedures are preferred, especially for design purposes or preliminary  
286 analysis of structures, a simplified Kelvin-Voigt model (spring and dashpot connected in parallel)  
287 can be used, the parameters of which can be determined based on the natural frequency of the  
288 structure and a design permitted damper displacement.

289  
290

### 291 **Modified Generalized Maxwell Model**

292 In trying to fit the GMM to the experimental data, it was found that the GMM was only  
293 adequate to describe VE behavior, regardless of the number of Maxwell elements used, and could  
294 not adequately represent the frequency or amplitude dependence. Hence, a Modified Generalized

295 Maxwell Model (MGMM) is proposed here. In order to take into account the non linear part of the  
 296 elastomer's behavior, a non linear damper was added to the main GMM equation:

$$297 \quad F_{NL} = c_{NL}|\dot{u}|^\alpha \text{sgn}(\dot{u}) \quad (10)$$

298 It was also found that the ED's behavior depended on both the maximum displacement and  
 299 the maximum velocity obtained during the loading history. Karavasilis et al. (2012) proposed a  
 300 modified stiffness based on the maximum displacement to take into account the softening of the  
 301 elastomer:

$$302 \quad k_{mod} = k_a e^{-\frac{u_{max}}{u_{ref}}} + k_b \quad (11)$$

303 where  $k_a$ ,  $k_b$  and  $u_{ref}$  are constants, and  $u_{max}$  is the average of the maximum absolute deformation  
 304 amplitudes in the negative ( $u_{max,n}$ ) and positive ( $u_{max,p}$ ) directions, and can be determined as  
 305 follows:

$$306 \quad u_{max} = \frac{u_{max,p} + |u_{max,n}|}{2} \quad (12)$$

307 The values of  $u_{max,p}$  and  $u_{max,n}$  are updated at every time step and are fed into the constitutive  
 308 equation to calculate the overall response. Similar modeling approach was followed by Summers  
 309 (2004). Following the same philosophy in terms of velocity instead of displacement, a modified  
 310 damping coefficient is introduced here, which was found to significantly contribute to the elastomer's  
 311 dynamic behavior:

$$312 \quad c_{mod} = c_a e^{-\frac{v_{max}}{v_{ref}}} + c_b \quad (13)$$

313 The parameters  $c_a$ ,  $c_b$ ,  $v_{ref}$ , and  $v_{max}$  are defined in a similar way to the terms in  $k_{mod}$ , of Eq. 11.

314 Hence, the final form of the Modified Generalized Maxwell Model (MGMM) consists of the

315 following parts:

- 316 • The linear GMM obtained from Equation 9
- 317 • The final version of  $F_0$ , which consists of a linear spring:

$$318 \quad F_0 = k_0 u \quad (14)$$

- 319 • The non linear damper, Eq. 10
- 320 • The modified stiffness, Eq. 11
- 321 • The modified damping coefficient, Eq. 13

322 Since a generalized equation for  $N$  Maxwell elements was now available, efforts were made  
323 to use more than one element. Analytical models were determined with  $N = 2, 3$  and 4 Maxwell  
324 elements. However, it was found that using only one Maxwell element was adequate for the specific  
325 material without significant compromise in accuracy. Hence, using  $N = 1$ , Eq. 9 simplifies to:

$$326 \quad F + \frac{c_1}{k_1} \dot{F} = F_0 + \frac{c_1}{k_1} \dot{F}_0 + c_1 \dot{u} \quad (15)$$

327 where  $\dot{F}$  and  $\dot{u}$  denote the differentiation of the force and the displacement with respect to time.

328 Therefore, incorporating the additional non-linear terms, the final constitutive equation for the  
329 dynamic behavior of the elastomeric dampers is:

$$330 \quad F + \frac{c_1}{k_1} \dot{F} = (k_0 + k_{mod})u + \left(\frac{c_1}{k_1} k_0 + c_1 + c_{mod}\right)\dot{u} + c_{NL} |\dot{u}|^\alpha \text{sgn}(\dot{u}) \quad (16)$$

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332

333

## Parameter Estimation

Least square minimization is one of the most widely used methods (Fan; Lee; Karavasilis et al.) of fitting the numerical parameters of a proposed model into the experimental data. However, as mentioned earlier, in case of the GMM this minimization technique has been based on the frequency domain, trying to minimize the shear storage modulus (from Eq.6), and the loss factor,  $n = \frac{G''}{G'}$  (combining Eq.6, and Eq.7) for every frequency and temperature that the damper was experimentally tested, as:

$$\min \left\{ \sum_{j=1}^N \left[ G'_{anal} - G'_{exp} \right]^2 \right\} \quad (17)$$

$$\min \left\{ \sum_{j=1}^N \left[ n_{anal} - n_{exp} \right]^2 \right\} \quad (18)$$

where  $N$  denotes the number of experimental data available. With regard to fitting the numerical model to the experimental data, researchers have used various techniques. The most common is to assume constant one of the three main parameters that the material is dependent upon (strain amplitude, loading frequency, ambient temperature) (Fan 1998). On the other hand, minimization of the corresponding error for every cyclic test has been used as well (Karavasilis et al. 2012), taking the average value of these parameters in order to generate the model through the whole range of shear strain amplitudes, frequencies, and temperatures. Finally, another widely used method is to include all the available data into one least square equation (Lee 2003).

In the case of models which were minimized based on their shear storage modulus, and loss factor for specific frequencies, even though the numerical values of  $G'$  and  $n$  seem to fit well with the experimental data, the corresponding model will not necessarily be valid in the time domain for random loading (and more specifically for random displacements), since these parameters will only be adequate to predict the shear storage modulus, and the loss factor of the damper under cyclic testing with constant frequency, and constant maximum displacement, and not the actual

358 elastomer's dynamic behavior.

359 Another approach was therefore used here, which takes into account the variability of displace-  
360 ments. Sweep amplitude sinusoidal tests (with the shear strain amplitude ranging from 10%-50%),  
361 which were carried out for the same frequency range used in the characterization tests, formed the  
362 basis for the minimization. The sweep amplitude tests were based on proposals from Dorka and  
363 Garcia (2005). Keeping the frequency and the temperature constant within each test, and setting the  
364 maximum strain amplitude to be 50%, the sweep amplitude tests consisted of the following steps:  
365 1.5 cycles of  $u_{max}/2$ , 1.5 cycles of  $u_{max}$ , 1.5 cycles of  $u_{max}/3$ , 1.5 cycles of  $u_{max}/6$ , 1.5 cycles of  
366  $u_{max}/2$ , 1.5 cycles of  $2u_{max}/3$ , 1.5 cycles of  $5u_{max}/6$ , 3 cycles of  $u_{max}$ , 1.5 cycles of  $5u_{max}/6$ , 1.5  
367 cycles of  $2u_{max}/3$ , 1.5 cycles of  $u_{max}/2$ , 1.5 cycles of  $u_{max}/3$ , 1.5 cycles of  $u_{max}/6$ . Fig. 8 shows  
368 a typical example of both the sweep sinusoidal command displacement and the ED's response for  
369 a test at a frequency of 3.0 Hz.

370 The sweep amplitude test is believed to be a more representative test compared with tests based  
371 on fixed strain amplitude and frequency, at least for fitting the parameters because of the strong  
372 amplitude dependence of the elastomer. The MGMM (Eq. 16) contains 11 parameters which need  
373 to be determined:  $k_1$ ,  $c_1$ ,  $k_0$ ,  $k_a$ ,  $k_b$ ,  $u_{ref}$ ,  $c_a$ ,  $c_b$ ,  $v_{ref}$ ,  $c_{NL}$ ,  $\alpha$ . In order to optimize these parameters  
374 a least square method was used:

$$375 \min \left\{ \sum_i \sum_k \left[ F_{anal(i,k)} - F_{exp(i,k)} \right]^2 \right\} \quad (19)$$

376 where  $i$  is the index for frequency content (e.g.,  $i = 2$  is equivalent with frequency of 0.5 Hz), and  
377  $k$  stands for the force-displacement data set at a specific frequency and displacement. In order to  
378 separate strain rate dependent and independent components of the model, stress relaxation tests  
379 along with very low frequency (0.01 Hz) cyclic tests in a range of strain amplitudes were carried  
380 out. These experiments allowed the separation of the strain rate dependent and independent parts  
381 of the stress. And therefore, quantifying the strain rate dependent damper's behaviour could be  
382 achieved due to these tests.

383 Table 3 provides the parameters of the hysteresis model, while Fig. 9 shows the hysteresis  
384 response of the damper for both the analytical and experimental case. It can be seen that the results  
385 are in a very good agreement. It is also worth mentioning that the proposed model is valid under  
386 any random loading, and its parameters do not depend on the imposed displacement or frequency.

387 In order to evaluate the effectiveness of GMM as well, experimental data of 0.5 and 2.0 Hz were  
388 compared with a GMM. Parameters of the GMM were obtained following the exact same process  
389 which was followed with MGMM. Fig. 10 shows this comparison, where the NRMS error for the  
390 data of 0.5 and 2.0 Hz is 8.1% and 2.43% respectively. Two things worth mentioning based on the  
391 GMM model. First, that even though GMM can acceptably capture the behavior of the material  
392 when the frequency of the loading remains constant, this is not the case when the frequency changes,  
393 even slightly. Second, as it can be seen from these figures, comparing simulation and experimental  
394 results in Force-Time relationship may be misleading, as it was mentioned earlier as well. Figs.  
395 10a, and 10b show that what seems like a good agreement in Force-Time relationship, is not really  
396 acceptable as a solution when it is compared in Force-Displacement terms.

### 397 **Effect of Temperature**

398 In order to capture the behavior of the EDs under different ambient temperatures, both the  
399 characterization tests and the sweep amplitude tests were repeated at different temperatures: 25,  
400 30, and 35°C. The same model (MGMM), which was developed and validated for the experiments  
401 which were carried out at room temperature (20°C), will be used here as well, by introducing a  
402 new parameter,  $\gamma_T$ . A simplifying assumption that all parameters change at the same rate with  
403 temperature was made. This assumption was considered as a simplifying tool when optimizing the  
404 parameters of the model under different temperatures, and can be seen as an additional empirical  
405 value. This led to a much more adjustable model, according to which the only parameter that  
406 needs to be specified when different ambient temperatures are applied, is the proposed parameter,  
407  $\gamma_T$ . Alternatively, additional optimization techniques identical with the ones carried out for room  
408 temperature, have to be used in the 25-35°C range in order to re-optimize the parameters of the  
409 model under the new temperatures. This alternative solution was also carried out and it was found

410 out that the values of the new parameters were almost the same as the original ones multiplied by  
 411  $\gamma_T$ . Hence, the first approach was adapted here. The comparison of experimental and simulation  
 412 data led to a very good agreement. The original proposed hysteresis model in this case has been  
 413 adjusted:

$$F_T + \frac{c_1}{k_1} \dot{F}_T = (\gamma_T k_0 + k_{mod,T})u + \left(\frac{c_1}{k_1} k_0 \gamma_T + \gamma_T c_1 + c_{mod,T}\right)\dot{u} + \gamma_T k_{NL}(\dot{u})^{\alpha \gamma_T} \quad (20)$$

414 where:

$$c_{mod,T} = \gamma_T c_a e^{-\frac{v_{max}}{\gamma_T v_{ref}}} + \gamma_T c_b \quad (21)$$

416 and

$$k_{mod,T} = \gamma_T k_a e^{-\frac{u_{max}}{\gamma_T u_{ref}}} + \gamma_T k_b \quad (22)$$

418 where T denotes the ambient temperature, and  $\gamma_T$  the coefficient which controls the parameters of  
 419 the model in different temperatures. Again, the same minimization techniques introduced earlier  
 420 were used to determine the  $\gamma_T$  parameter for different temperatures. These values are summarized  
 421 in Table 4.

422 As expected from the results of the characterization tests carried out at different temperatures,  
 423 the values of  $\gamma_T$  reduce with the increase of temperature. This reduction can be schematically seen  
 424 at Fig. 11 and can be approximately determined as 0.116/°C. A more quantified approach is given  
 425 in Table 5 where the Normalized Root Mean Square (NRMS) error has been calculated for every  
 426 sweep amplitude test which was carried out under the aforementioned ranges of frequencies and  
 427 temperatures. It can be seen that the the proposed constitutive model can accurately predict the  
 428 response of the elastomer under any frequency, strain amplitude or ambient temperature, since the  
 429 maximum NRMS error observed during these tests was 1.83%.

430 It should be noted that this model takes into account the ambient temperature during the test,  
431 and not the temperature due to energy dissipation of the damper. Due to the limited number of  
432 cycles used in this study, the temperature rise during the tests was not found to be dominant factor  
433 to the proposed model. However, if the material is tested under more cycles, for validation against  
434 wind loading for example, then this alteration of temperature should be taken into account.

435 Moreover, in order to evaluate the efficiency of the proposed model, in terms of dissipative  
436 energy, Figs 12 and 13 can be used which compare the energy dissipation extracted from the model  
437 with the dissipation energy obtained from the characterization tests, for  $T=20C$  and  $T= 30C$ , for  
438 every strain amplitude and frequency. It can be seen that dissipation energy is another parameter  
439 which shows the efficiency of the proposed model.

## 440 **CONCLUSIONS**

441 A constitutive equation, which is able to capture the dynamic characteristics of an elastomeric  
442 damper, is proposed in this study. The properties of the damper were first measured experimentally  
443 over a wide range of frequencies, amplitudes and temperatures. It was seen that strain amplitude  
444 had the largest impact on the EDs behavior. It was also concluded that the loss factor of the material  
445 is almost constant when frequency changes. The most noticeable remark though, is the change of  
446 elastomer's behavior, which can be only captured by comparing hysteresis behavior, when the strain  
447 amplitude increases above 30%. Therefore, not only the mechanical parameters of the material  
448 change, but also the hysteresis shape itself. It was shown that when the material has larger strain  
449 amplitudes it behaves more like a typical elastomer, and tends to be more and more independent of  
450 the frequency, in contrast with its viscoelastic phase when the strain amplitudes are relatively low.

451 In order to capture this behavior a new constitutive model was presented, based on a modified  
452 version of GMM. It was found that, for our system, a GMM with a single spring-damper ( $N = 1$ ) was  
453 sufficient, with the addition of further elements giving only modest improvements in performance.  
454 However, it was then necessary to modify this linear model through the addition of amplitude-related  
455 adjustments to stiffness and damping terms, and a further non-linear damping term. The model was  
456 then extended to include a simple adjustment to account for variations in ambient temperature. The

457 material parameters of this model have been identified using experimental results. The resulting  
458 model, termed the Modified GMM, shows excellent agreement with the experimental results across  
459 a wide range of operating conditions.

#### 460 **DATA AVAILABILITY STATEMENT**

461 All data, models, and code generated or used during the study appear in the submitted article.

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466 **APPENDIX I. FORCE-DISPLACEMENT RELATIONSHIP OF GENERALIZED MAXWELL**  
 467 **MODEL**

468 As already mentioned the Generalized Maxwell Model (GMM) is a series of Maxwell elements  
 469 assembled in parallel along with a linear spring (Fig. 7). For every Maxwell element the force  
 470 displacement relationship in time domain is given as:

$$471 \quad k_i F_i + c_i \frac{dF_i}{dt} = k_i c_i \frac{du_i}{dt} \quad (23)$$

472 where  $k_i$  and  $c_i$  are the stiffness and the dashpot coefficients of the  $i$ th Maxwell element,  $u_i$  is  
 473 the imposed displacement and  $F_i$  the corresponding force. In the case of parallel connection of  
 474 Maxwell elements  $u_i = u$  for every,  $i$ , Maxwell element. Assuming zero initial conditions for both  
 475 the displacement and the force, and applying Laplace Transformation to Eq.23:

$$476 \quad (k_i + c_i s) F_{i,L} = k_i c_i s u_L \quad (24)$$

477 where  $F_{i,L}$  is the Laplace transform of the force corresponding to the  $i$  Maxwell element, and  $u_L$  is  
 478 the Laplace transform of the displacement. Equation 24 can be rewritten in the following form:

$$479 \quad F_{i,L} = \frac{k_i c_i s}{k_i + c_i s} u_L \quad (25)$$

480 And for  $N$  Maxwell elements the Laplace transform of the total force can be determined as:

$$481 \quad F_L = \sum_{i=1}^N F_{i,L} + F_{0,L} = \sum_{i=1}^N \frac{k_i c_i s}{k_i + c_i s} u_L + F_{0,L} \quad (26)$$

482 where  $F_{0,L}$  represents the Laplace transform of the additional force corresponding to the linear  
 483 spring (in this case), which is connected in parallel with the  $N$  Maxwell elements. Multiplying  
 484 throughout by the minimum common multiple of the denominators of the operators of the right

485 side of Equation 26 and rearranging:

$$486 \quad \left[ \prod_{h=1}^N (k_h + c_h s) \right] F_L = \sum_{i=1}^N \left[ (k_i c_i s) \prod_{j=1}^N (k_j + c_j s) \right] u_L + \left[ \prod_{g=1}^N (k_g + c_g s) \right] F_{0,L} \quad (27)$$

487 where  $j \neq i$ ,  $\Sigma$  represents summation, and  $\Pi$  represents a series multiplication

Focusing only on the left part of the Equation 27:

$$\begin{aligned} & \left[ \prod_{h=1}^N (k_h + c_h s) \right] F_L = \\ & \left[ (k_1 + c_1 s)(k_2 + c_2 s)(k_3 + c_3 s) \dots (k_{N-1} + c_{N-1} s)(k_N + c_N s) \right] F_L = \\ & \left[ (k_1 k_2 k_3 \dots k_{N-2} k_{N-1} k_N) + \right. \\ & \quad (k_1 k_2 k_3 \dots k_{N-2} k_{N-1} c_N + k_1 k_2 k_3 \dots k_{N-2} k_N c_{N-1} + \dots + \\ & \quad \left. k_1 k_3 \dots k_{N-2} k_{N-1} k_N c_2 + k_2 k_3 \dots k_{N-2} k_{N-1} k_N c_1) s + \right. \\ & \quad (k_1 k_2 k_3 \dots k_{N-2} c_{N-1} c_N + k_1 k_2 k_3 \dots k_{N-4} k_{N-3} k_{N-1} c_{N-2} c_N + \dots + \\ & \quad \left. k_1 k_4 k_5 \dots k_{N-2} k_{N-1} k_N c_2 c_3 + k_3 k_4 k_5 \dots k_{N-2} k_{N-1} k_N c_1 c_2) s^2 + \right. \\ & \quad (k_1 k_2 k_3 \dots k_{N-4} k_{N-3} c_{N-2} c_{N-1} c_N + k_1 k_2 k_3 \dots k_{N-4} k_{N-2} c_{N-3} c_{N-1} c_N + \dots + \\ & \quad \left. k_1 k_5 k_6 \dots k_{N-1} k_N c_2 c_3 c_4 + k_4 k_5 k_6 \dots k_{N-1} k_N c_1 c_2 c_3) s^3 \right. \\ & \quad + \dots + \\ & \quad (k_1 c_2 c_3 \dots c_{N-2} c_{N-1} c_N + k_2 c_1 c_3 \dots c_{N-2} c_{N-1} c_N + \dots + \\ & \quad \left. k_{N-1} c_1 c_2 \dots c_{N-3} c_{N-2} c_N + k_N c_1 c_2 \dots c_{N-2} c_{N-1}) s^{N-1} + \right. \\ & \quad \left. (c_1 c_2 c_3 \dots c_{N-2} c_{N-1} c_N) s^N \right] F_L \quad (28) \end{aligned}$$

Dividing Equation 28 by  $\prod_{d=1}^N (k_d)$ :

$$\begin{aligned}
& \left[ 1 + \left( \frac{c_N}{k_N} + \frac{c_{N-1}}{k_{N-1}} + \dots + \frac{c_2}{k_2} + \frac{c_1}{k_1} \right) s + \right. \\
& \left( \frac{c_N}{k_N} \frac{c_{N-1}}{k_{N-1}} + \frac{c_N}{k_N} \frac{c_{N-2}}{k_{N-2}} + \frac{c_N}{k_N} \frac{c_{N-3}}{k_{N-3}} + \dots + \frac{c_2}{k_2} \frac{c_3}{k_3} + \frac{c_1}{k_1} \frac{c_2}{k_2} \right) s^2 + \\
& \left( \frac{c_N}{k_N} \frac{c_{N-1}}{k_{N-1}} \frac{c_{N-2}}{k_{N-2}} + \frac{c_N}{k_N} \frac{c_{N-1}}{k_{N-1}} \frac{c_{N-3}}{k_{N-3}} + \dots + \frac{c_2}{k_2} \frac{c_3}{k_3} \frac{c_4}{k_4} + \frac{c_1}{k_1} \frac{c_2}{k_2} \frac{c_3}{k_3} \right) s^3 \\
& + \dots + \\
& \left. \left( \frac{c_2}{k_2} \frac{c_3}{k_3} \dots \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} + \frac{c_1}{k_1} \frac{c_3}{k_3} \dots \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} + \dots + \frac{c_1}{k_1} \frac{c_2}{k_2} \dots \frac{c_{N-3}}{k_{N-3}} \frac{c_{N-2}}{k_{N-2}} \frac{c_N}{k_N} + \frac{c_1}{k_1} \frac{c_2}{k_2} \dots \frac{c_{N-2}}{k_{N-2}} \frac{c_{N-1}}{k_{N-1}} \right) s^{N-1} + \right. \\
& \left. \left( \frac{c_1}{k_1} \frac{c_2}{k_2} \dots \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} \right) s^N \right] F_L
\end{aligned} \tag{29}$$

488 Equation 29 can be rewritten in the following format:

$$F_L + \sum_{a=1}^N \left[ s^a F_L \left\{ \sum_{m_i \in A_i} \left\{ \prod_{r \in B} \left( \frac{c_r}{k_r} \right) \right\} \right\} \right] \tag{30}$$

490 where:

- 491 •  $i \in \{1, 2, \dots, \alpha\}$
- 492 •  $A_i \in \{i, i+1, i+2, \dots, N-\alpha+i\}$
- 493 •  $m_i > m_{i-1}$ , for  $i > 1$
- 494 •  $B \in \{m_1, m_2, \dots, m_{\alpha-1}, m_\alpha\}$

Focusing now on the first part of the right hand side of Equation 27:

$$\begin{aligned}
& \left[ \sum_{i=1}^N (k_i c_i s) \prod_{j=1}^N (k_j + c_j s) \right] u_L = \\
& \left[ k_1 c_1 s (k_2 + c_2 s) (k_3 + c_3 s) \dots (k_{N-1} + c_{N-1} s) (k_N + c_N s) + \right. \\
& k_2 c_2 s (k_1 + c_1 s) (k_3 + c_3 s) \dots (k_{N-1} + c_{N-1} s) (k_N + c_N s) + \\
& + \dots + \\
& k_{N-1} c_{N-1} s (k_1 + c_1 s) (k_2 + c_2 s) \dots (k_{N-2} + c_{N-2} s) (k_N + c_N s) + \\
& \left. k_N c_N s (k_1 + c_1 s) (k_2 + c_2 s) \dots (k_{N-2} + c_{N-2} s) (k_{N-1} + c_{N-1} s) \right] u_L
\end{aligned} \tag{31}$$

Equation 31 can be rewritten as:

$$\begin{aligned}
& \left[ \left[ (c_1 + c_2 + \dots + c_{N-1} + c_N) k_1 k_2 \dots k_{N-1} k_N \right] s + \right. \\
& \left[ c_1 c_2 (k_1 + k_2) k_3 \dots k_N + c_1 c_3 (k_1 + k_3) k_2 k_4 \dots k_N + \dots + \right. \\
& \quad \left. c_{N-1} c_N (k_{N-1} + k_N) k_1 k_2 \dots k_{N-2} \right] s^2 + \\
& \left[ c_1 c_2 c_3 (k_1 + k_2 + k_3) k_4 \dots k_N + c_1 c_2 c_4 (k_1 + k_2 + k_4) k_3 k_5 \dots k_N + \dots + \right. \\
& \quad \left. c_{N-2} c_{N-1} c_N (k_{N-2} + k_{N-1} + k_N) k_1 \dots k_{N-3} \right] s^3 \\
& \quad + \dots + \\
& \left[ c_1 c_2 \dots c_{N-2} c_{N-1} (k_1 + k_2 + \dots + k_{N-2} + k_{N-1}) k_N + \dots + \right. \\
& \quad \left. c_2 c_3 \dots c_{N-1} c_N (k_2 + k_3 \dots + k_{N-1} + k_N) k_1 \right] s^{N-1} + \\
& \left. \left[ c_1 c_2 \dots c_{N-1} c_N (k_1 + k_2 + \dots + k_{N-1} + k_N) \right] s^N \right] u_L
\end{aligned} \tag{32}$$

And now dividing Equation 32 by  $\prod_{d=1}^N (k_d)$ , as was done with Equation 28:

$$\begin{aligned}
& \left[ \left[ c_1 + c_2 + \dots + c_{N-1} + c_N \right] s + \right. \\
& \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} (k_1 + k_2) + \frac{c_1}{k_1} \frac{c_3}{k_3} (k_1 + k_3) + \dots + \right. \\
& \quad \left. \frac{c_{N-2}}{k_{N-2}} \frac{c_{N-1}}{k_{N-1}} (k_{N-2} + k_{N-1}) + \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} (k_{N-1} + k_N) \right] s^2 + \\
& \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \frac{c_3}{k_3} (k_1 + k_2 + k_3) + \frac{c_1}{k_1} \frac{c_2}{k_2} \frac{c_4}{k_4} (k_1 + k_2 + k_4) + \dots + \right. \\
& \quad \left. \frac{c_{N-2}}{k_{N-2}} \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} (k_{N-2} + k_{N-1} + k_N) \right] s^3 \\
& \quad + \dots + \\
& \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \dots \frac{c_{N-2}}{k_{N-2}} \frac{c_{N-1}}{k_{N-1}} (k_1 + k_2 + \dots + k_{N-2} + k_{N-1}) + \right. \\
& \quad \frac{c_1}{k_1} \frac{c_3}{k_3} \dots \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} (k_1 + k_3 + \dots + k_{N-1} + k_N) + \dots + \\
& \quad \left. \frac{c_2}{k_2} \frac{c_3}{k_3} \dots \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} (k_2 + k_3 + \dots + k_{N-1} + k_N) \right] s^{N-1} + \\
& \left. \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \dots \frac{c_{N-1}}{k_{N-1}} \frac{c_N}{k_N} (k_1 + k_2 + \dots + k_{N-1} + k_N) \right] s^N \right] u_L
\end{aligned} \tag{33}$$

Which leads to:

$$\sum_{\beta=1}^N \left[ s^\beta u_L \left\{ \sum_{x_c \in Q_c} \left\{ \prod_{z \in W} \left( \frac{c_z}{k_z} \right) \sum_{z \in W} k_z \right\} \right\} \right] \quad (34)$$

where:

- $c \in \{1, 2, \dots, \beta\}$
- $Q_c \in \{c, c+1, c+2, \dots, N-\beta+c\}$
- $x_c > x_{c-1}$  for  $c > 1$
- $W \in \{x_1, x_2, \dots, x_{\beta-1}, x_\beta\}$

The last part of the right hand side of the Equation 27 can be similarly rearranged with its left part and can be presented in its final format:

$$F_{0,L} + \sum_{d=1}^N \left[ s^d F_{0,L} \left\{ \sum_{p_v \in H_v} \left\{ \prod_{f \in L} \left( \frac{c_f}{k_f} \right) \right\} \right\} \right] \quad (35)$$

where:

- $v \in \{1, 2, \dots, d\}$
- $H_v \in \{v, v+1, v+2, \dots, N-d+v\}$
- $p_v > p_{v-1}$  for  $v > 1$
- $L \in \{p_1, p_2, \dots, p_{d-1}, p_d\}$

Combining Equations 30,34, and 35, then 27 can be rewritten as:

$$F_L + \sum_{a=1}^N \left[ s^a F_L \left\{ \sum_{m_i \in A_i} \left\{ \prod_{r \in B} \left( \frac{c_r}{k_r} \right) \right\} \right\} \right] = \sum_{\beta=1}^N \left[ s^\beta u_L \left\{ \sum_{x_c \in Q_c} \left\{ \prod_{z \in W} \left( \frac{c_z}{k_z} \right) \sum_{z \in W} k_z \right\} \right\} \right] + \left[ F_{0,L} + \sum_{d=1}^N \left[ s^d F_{0,L} \left\{ \sum_{p_v \in H_v} \left\{ \prod_{f \in L} \left( \frac{c_f}{k_f} \right) \right\} \right\} \right] \right] \quad (36)$$

Finally, applying inverse Laplace transformation to 36:

$$F + \sum_{a=1}^N \left[ \frac{\partial^a F}{\partial t^a} \left\{ \sum_{m_i \in A_i} \left\{ \prod_{r \in B} \left( \frac{c_r}{k_r} \right) \right\} \right\} \right] = \sum_{\beta=1}^N \left[ \frac{\partial^\beta u}{\partial t^\beta} \left\{ \sum_{x_c \in Q_c} \left\{ \prod_{z \in W} \left( \frac{c_z}{k_z} \right) \sum_{z \in W} k_z \right\} \right\} \right] + \left[ F_0 + \sum_{d=1}^N \left[ \frac{\partial^d F_0}{\partial t^d} \left\{ \sum_{p_v \in H_v} \left\{ \prod_{f \in L} \left( \frac{c_f}{k_f} \right) \right\} \right\} \right] \right] \quad (37)$$

510 where:

- 511 •  $i \in \{1, 2, \dots, \alpha\}$ ,  $c \in \{1, 2, \dots, \beta\}$ ,  $v \in \{1, 2, \dots, d\}$
- 512 •  $A_i \in \{i, i+1, i+2, \dots, N-\alpha+i\}$ ,  $Q_c \in \{c, c+1, c+2, \dots, N-\beta+c\}$ ,  $H_v \in \{v, v+1, v+2, \dots, N-d+v\}$
- 513 •  $B \in \{m_1, m_2, \dots, m_{\alpha-1}, m_\alpha\}$ ,  $W \in \{x_1, x_2, \dots, x_{\beta-1}, x_\beta\}$ , and  $L \in \{p_1, p_2, \dots, p_{d-1}, p_d\}$
- 514 •  $m_i > m_{i-1}$ ,  $x_c > x_{c-1}$ ,  $p_v > p_{v-1}$  for  $i, c, v > 1$

For illustrative purposes, the cases of  $N=1$ ,  $N=2$ , and  $N=3$  will be presented. If  $N=1$  Eq. 37 becomes:

$$F + \frac{dF}{dt} \left[ \frac{c_1}{k_1} \right] = F_0 + \frac{dF_0}{dt} \frac{c_1}{k_1} + \frac{du}{dt} c_1 \quad (38)$$

In the case of  $N=2$  Eq. 37 is determined as:

$$F + \frac{dF}{dt} \left[ \frac{c_1}{k_1} + \frac{c_2}{k_2} \right] + \frac{d^2 F}{dt^2} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \right] = F_0 + \frac{dF_0}{dt} \left[ \frac{c_1}{k_1} + \frac{c_2}{k_2} \right] + \frac{d^2 F_0}{dt^2} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \right] + \frac{du}{dt} [c_1 + c_2] + \frac{d^2 u}{dt^2} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} (k_1 + k_2) \right] \quad (39)$$

In the case of  $N=3$  Eq. 37 is determined as:

$$F + \frac{dF}{dt} \left[ \frac{c_1}{k_1} + \frac{c_2}{k_2} + \frac{c_3}{k_3} \right] + \frac{d^2 F}{dt^2} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} + \frac{c_1}{k_1} \frac{c_3}{k_3} + \frac{c_2}{k_2} \frac{c_3}{k_3} \right] + \frac{d^3 F}{dt^3} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \frac{c_3}{k_3} \right] = F_0 + \frac{dF_0}{dt} \left[ \frac{c_1}{k_1} + \frac{c_2}{k_2} + \frac{c_3}{k_3} \right] + \frac{d^2 F_0}{dt^2} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} + \frac{c_1}{k_1} \frac{c_3}{k_3} + \frac{c_2}{k_2} \frac{c_3}{k_3} \right] + \frac{d^3 F_0}{dt^3} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \frac{c_3}{k_3} \right] + \frac{du}{dt} [c_1 + c_2 + c_3] + \frac{d^2 u}{dt^2} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} (k_1 + k_2) + \frac{c_1}{k_1} \frac{c_3}{k_3} (k_1 + k_3) + \frac{c_2}{k_2} \frac{c_3}{k_3} (k_2 + k_3) \right] + \frac{d^3 u}{dt^3} \left[ \frac{c_1}{k_1} \frac{c_2}{k_2} \frac{c_3}{k_3} (k_1 + k_2 + k_3) \right] \quad (40)$$

## REFERENCES

- Aiken, I. (1997). "An analytical hysteresis model for elastomeric seismic isolation bearings." *Earthquake Engineering and Structural Dynamics*, 26(2).
- Bland, D. (1960). *The Theory of Linear Viscoelasticity*, Pergamon Press, Oxford, 1960.
- Constantinou, M. C., Soong, T. T., and Dargush, G. F. (1998). *Passive energy dissipation systems for structural design and retrofit*. Multidisciplinary Center for Earthquake Engineering Research Buffalo, New York.
- Constantinou, M. C. and Tadjbakhsh, I. G. (1985). "Hysteretic dampers in base isolation: random approach." *Journal of Structural Engineering*, 111(4), 705–721.
- Dahl, P. R. (1968). *A solid friction model*. Aerospace Corp El Segundo Ca.
- Dorka, U. and Garcia, J. (2005). "Seismic qualification of passive mitigation devices." *Report no.1,2005*, Cooperative Advancements in Seismic and Dynamic Experiments (CASCADE).
- Fan, C. P. (1998). *Seismic Analysis, Behavior, and Retrofit of Non-Ductile Reinforced Concrete Frame Buildings with Viscoelastic Dampers*. Ph.D. Dissertation, Department of Civil and Environmental Engineering, Lehigh University.
- Ferry, J. D. (1980). *Viscoelastic properties of polymers*. John Wiley & Sons.
- Hepburn, C. and Reynolds, R. (1979). *Elastomers: Criteria for Engineering Design*. Applied Science Publishers, London.
- Hwang, J. and Wang, J. (1998). "Seismic response prediction of hbr bearings using fractional derivative maxwell model." *Engineering Structures*, 20(9), 849–856.
- Jrad, H., Dion, J. L., Renaud, F., Tawfiq, I., and Haddar, M. (2013). "Experimental characterization, modeling and parametric identification of the non linear dynamic behavior of viscoelastic components." *European Journal of Mechanics - A/Solids*, 42, 176 – 187.
- Jrad, H., Dion, J. L., Renaud, F., Tawfiq, I., and Haddar, M. (2017). "Experimental and numerical investigation of energy dissipation in elastomeric rotational joint under harmonic loading." *Mechanics of Time-Dependent Materials*, 21(2), 177–198.
- Karavasilis, T. L., Blakeborough, T., and Williams, M. S. (2011). "Development of nonlinear

542 analytical model and seismic analyses of a steel frame with self-centering devices and viscoelastic  
543 dampers.” *Computers & Structures*, 89(11), 1232–1240.

544 Karavasilis, T. L., Sause, R., and Ricles, J. M. (2012). “Seismic design and evaluation of steel  
545 moment-resisting frames with compressed elastomer dampers.” *Earthquake Engineering &  
546 Structural Dynamics*, 41(3), 411–429.

547 Kit Miyamoto, M. and Lon M. Determan, S. “Structural Applications of Taylor  
548 Fluid Viscous Dampers, <[http://www.taylordevices.com/custom/pdf/tech-papers/  
549 70-SeismicRehabilitationofHCS.pdf](http://www.taylordevices.com/custom/pdf/tech-papers/70-SeismicRehabilitationofHCS.pdf)>.

550 Koh, C. G. and Kelly, J. M. (1990). “Application of fractional derivatives to seismic analysis of  
551 base-isolated models.” *Earthquake engineering & structural dynamics*, 19(2), 229–241.

552 Kästner, M., Obst, M., Brummund, J., Thielsch, K., and Ulbricht, V. (2012). “Inelastic material  
553 behavior of polymers – experimental characterization, formulation and implementation of a  
554 material model.” *Mechanics of Materials*, 52, 40 – 57.

555 Lai, M., Lu, P., Lunsford, D., Kasai, K., and Chang, K. (1996). “Viscoelastic damper: a damper with  
556 linear or nonlinear material.” *Proceedings of 11th World Conference on Earthquake Engineering*.

557 Lee, K. S. (2003). *Seismic Behavior of Structures with Dampers Made from Ultra High Damping  
558 Natural Rubber*. Ph.D. Dissertation, Department of Civil and Environmental Engineering, Lehigh  
559 University.

560 Mahmoodi, P., Robertson, L., Yontar, M., Moy, C., and Feld, L. (1987). “Performance of viscoelastic  
561 dampers in world trade center towers.” *Dynamics of structures*, ASCE, 632–644.

562 Makris, N. and Constantinou, M. (1991). “Fractional-derivative maxwell model for viscous  
563 dampers.” *Journal of Structural Engineering*, 117(9), 2708–2724.

564 McCrum, N. G., Buckley, C., and Bucknall, C. B. (1997). *Principles of polymer engineering*.  
565 Oxford University Press, USA.

566 Muhr, A. (22 October 2019). *Personal Communication with TARRC*.

567 Nashif, A. D., Jones, D. I., and Henderson, J. P. (1985). *Vibration damping*. John Wiley & Sons.

568 Pant, D., Montgomery, M., Christopoulos, C., and Poon, D. (2017). “Viscoelastic coupling dampers

569 for the enhanced seismic resilience of a megatall building.” *16th World Conference on Earthquake*  
570 *Engineering, 16WCEE 2017*, Vol. 1318.

571 Papoulia, K. D. and Kelly, J. M. (1994). *Material characterization of elastomers used in earthquake*  
572 *base isolation*. Earthquake Engineering Research Center, University of California.

573 Petrone, F., Lacagnina, M., and Scionti, M. (2004). “Dynamic characterization of elastomers and  
574 identification with rheological models.” *Journal of Sound and Vibration*, 271(1), 339 – 363.

575 Prestandard (2000). “commentary for the seismic rehabilitation of buildings (fema356).” *Washing-*  
576 *ton, DC: Federal Emergency Management Agency*, 7.

577 Sause, R., Lee, K.-S., and Ricles, J. (2007). “Rate-independent and rate-dependent models for  
578 hysteretic behavior of elastomers.” *Journal of Engineering Mechanics*, 133(11), 1162–1170.

579 Shen, K. and Soong, T. (1995). “Modeling of viscoelastic dampers for structural applications.”  
580 *Journal of Engineering Mechanics*, 121(6), 694–701.

581 Shrimali, M., Bharti, S., and Dumne, S. (2015). “Seismic response analysis of coupled building  
582 involving mr damper and elastomeric base isolation.” *Ain Shams Engineering Journal*, 6(2), 457  
583 – 470.

584 Silwal, B., Michael, R. J., and Ozbulut, O. E. (2015). “A superelastic viscous damper for enhanced  
585 seismic performance of steel moment frames.” *Engineering Structures*, 105, 152 – 164.

586 Soong, T. T. and Dargush, G. F. (1997). *Passive energy dissipation systems in structural engineering*.  
587 Wiley.

588 Summers, P., Jacob, P., Marti, J., Bergamo, G., Dorfmann, L., Castellano, G., Poggianti, A.,  
589 Karabalis, D., Silbe, H., and Triantafillou, S. (2004). “Development of new base isolation  
590 devices for application at refineries and petrochemical facilities.” *13th World Conference on*  
591 *Earthquake Engineering, Vancouver, BC, Canada*, 1–6.

592 Taniwangsa, W. and Kelly, J. (1996). “Studies on seismic isolation for housing in developing  
593 regions.” *Proceedings of the 11th World Conference on Earthquake Engineering, Acapulco,*  
594 *Mexico*.

595 Teramoto, T., Kitamura, H., and Ozaki, H. (1996). “Practical application of high-damping rubber

- 596 dampers to a slender building.” *11th World Conference on Earthquake Engineering, 11WCEE*  
597 *1996*, Vol. 1801.
- 598 Torvik, P. and Bagley, R. (1984). “On the appearance of the fractional derivative in the behavior of  
599 real materials..” *Journal of Applied Mechanics, Transactions ASME*, 51(2), 294–298.
- 600 Vargas, R. and Bruneau, M. (2007). “Effect of supplemental viscous damping on the seismic re-  
601 sponse of structural systems with metallic dampers.” *Journal of Structural Engineering*, 133(10),  
602 1434–1444.
- 603 Yu, Y., Li, Y., Li, J., and Gu, X. (2016). “A hysteresis model for dynamic behaviour of magnetorhe-  
604 ological elastomer base isolator.” *Smart Materials and Structures*, 25(5), 055029.

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**TABLE 1.** Elastomer' Properties (Muhr 2019)

<b>Property</b>	<b>Unaged</b>	<b>Aged</b>
Dynamic Shear modulus (100% strain, 1Hz; MPa)	0.76	0.86
Dynamic Shear loss factor (100% strain, 1Hz)	0.52	0.5
Shear failure strain (strain rate $1\text{s}^{-1}$ ; %)	450	-
Tensile strength (MPa)	6.6	7.5
Tensile elongation at break	640	630
Trouser tear strength (N/mm)	6.7	-

**TABLE 2.** Mechanical Properties at a range of strain amplitudes, frequencies and temperatures

Temperatures	Shear Strain %	f=0.25Hz		f=0.5Hz		f=1Hz		f=2Hz		f=3Hz		f=4Hz	
		G'(MPa)	n	G'(MPa)	n	G'(MPa)	n	G'(MPa)	n	G'(MPa)	n	G'(MPa)	n
T =20°C	10	1.36	0.41	1.36	0.41	1.47	0.41	1.62	0.40	1.70	0.40	1.77	0.39
	20	1.04	0.36	1.08	0.35	1.15	0.35	1.24	0.35	1.30	0.35	1.34	0.35
	30	0.88	0.33	0.92	0.33	0.98	0.33	1.05	0.34	1.09	0.33	1.12	0.33
	40	0.81	0.34	0.84	0.33	0.89	0.33	0.95	0.33	0.98	0.33	1.01	0.33
	50	0.77	0.30	0.80	0.30	0.84	0.31	0.89	0.31	0.92	0.31	0.95	0.31
T =25°C	10	1.31	0.39	1.33	0.38	1.43	0.37	1.55	0.37	1.62	0.36	1.66	0.36
	20	1.01	0.33	1.01	0.33	1.08	0.33	1.16	0.33	1.21	0.32	1.25	0.32
	30	0.83	0.32	0.86	0.31	0.91	0.31	0.97	0.31	1.01	0.31	1.04	0.31
	40	0.75	0.30	0.77	0.30	0.82	0.30	0.86	0.31	0.89	0.30	0.92	0.30
	50	0.70	0.30	0.72	0.29	0.75	0.30	0.80	0.30	0.82	0.30	0.84	0.30
T =30°C	10	1.13	0.36	1.13	0.37	1.21	0.36	1.31	0.36	1.37	0.35	1.42	0.36
	20	0.88	0.32	0.89	0.32	0.95	0.32	1.02	0.32	1.07	0.32	1.10	0.31
	30	0.76	0.31	0.80	0.30	0.84	0.30	0.89	0.30	0.93	0.30	0.95	0.30
	40	0.67	0.29	0.71	0.28	0.75	0.29	0.79	0.29	0.82	0.29	0.84	0.29
	50	0.66	0.28	0.68	0.28	0.71	0.28	0.75	0.29	0.78	0.29	0.82	0.28
T =35°C	10	1.01	0.35	1.02	0.35	1.08	0.35	1.17	0.35	1.22	0.35	1.27	0.34
	20	0.80	0.30	0.82	0.30	0.87	0.31	0.94	0.31	0.97	0.30	1.00	0.30
	30	0.73	0.28	0.77	0.29	0.81	0.29	0.81	0.29	0.85	0.29	0.86	0.29
	40	0.64	0.28	0.67	0.28	0.70	0.28	0.75	0.29	0.77	0.28	0.79	0.28
	50	0.61	0.27	0.63	0.27	0.66	0.27	0.70	0.27	0.72	0.27	0.73	0.27

**TABLE 3.** MGMM Parameters under 20°C

<b>k1 (kN/mm)</b>	<b>c1 (kNsec/mm)</b>	<b>k0 (N/mm)</b>	<b>c_NL(kN/((mm/sec)<sup>0.374</sup>))</b>	<b><math>\alpha</math></b>	
0.515	0.0151	0.107	1.585	0.374	
<b>k_a (kN/mm)</b>	<b>k_b (kN/mm)</b>	<b>u_ref (mm)</b>	<b>c_a (kNsec/mm)</b>	<b>c_b (kNsec/mm)</b>	<b>v_ref (mm/sec)</b>
3.741	3.613	7.966	0.902	0.190	13.468

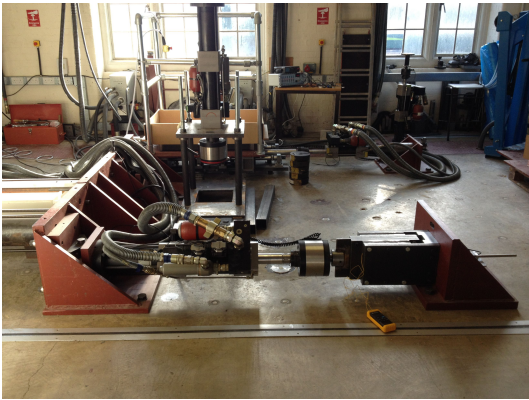
**TABLE 4.** Values of  $\gamma_T$  for different ambient temperatures

$\gamma_{20}$	$\gamma_{25}$	$\gamma_{30}$	$\gamma_{35}$
1	0.935	0.875	0.820

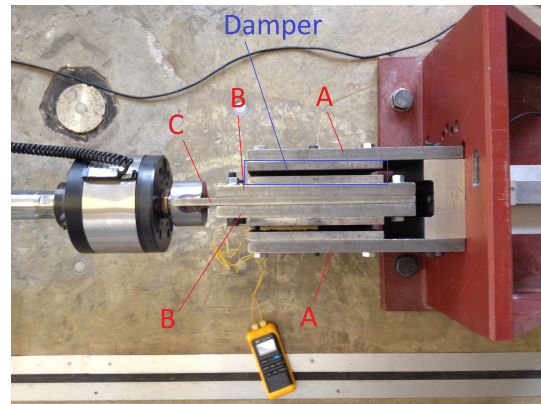
**TABLE 5.** NRMS error (%) between MGMM and experimental data for sweep amplitude tests under a range of frequencies and temperatures

	<b>Frequency (Hz)</b>					
<b>Temperature (C )</b>	<b>0.25</b>	<b>0.5</b>	<b>1</b>	<b>2</b>	<b>3</b>	<b>4</b>
<b>20</b>	1.60	1.45	1.72	1.73	1.41	1.57
<b>25</b>	1.64	1.51	1.76	1.68	1.71	1.83
<b>30</b>	1.10	1.24	1.57	1.49	1.51	1.55
<b>35</b>	1.46	1.61	1.52	1.72	1.64	1.59

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(a)

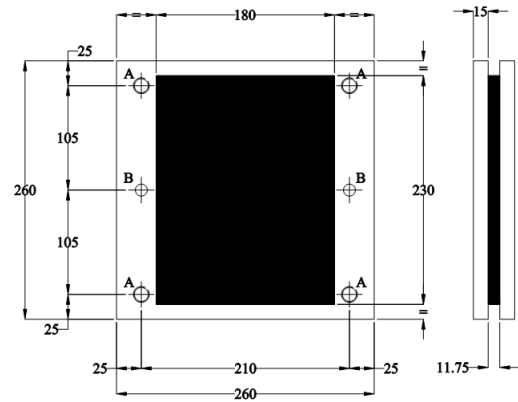


(b)

**Fig. 1.** Experimental Rig



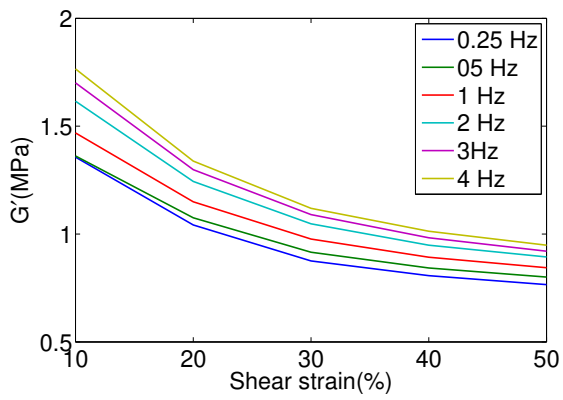
(a)



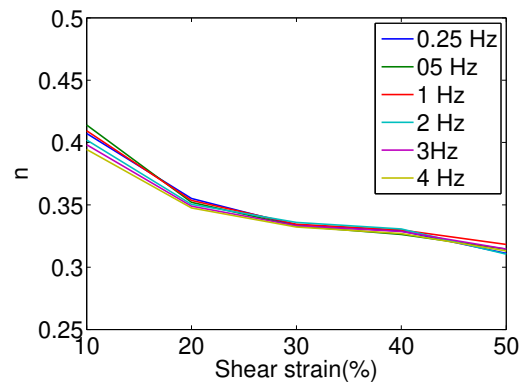
A - 4 HOLES M16 THRU  
B - 2 HOLES ø12 THRU

(b)

**Fig. 2.** Individual Damper

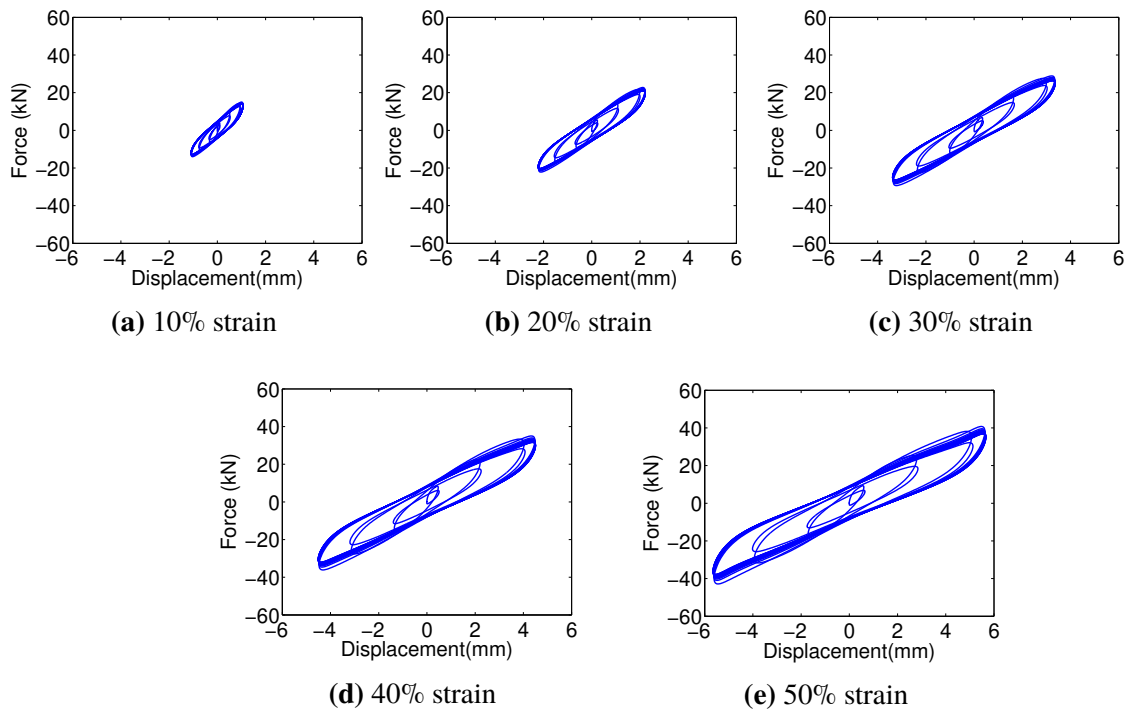


(a)

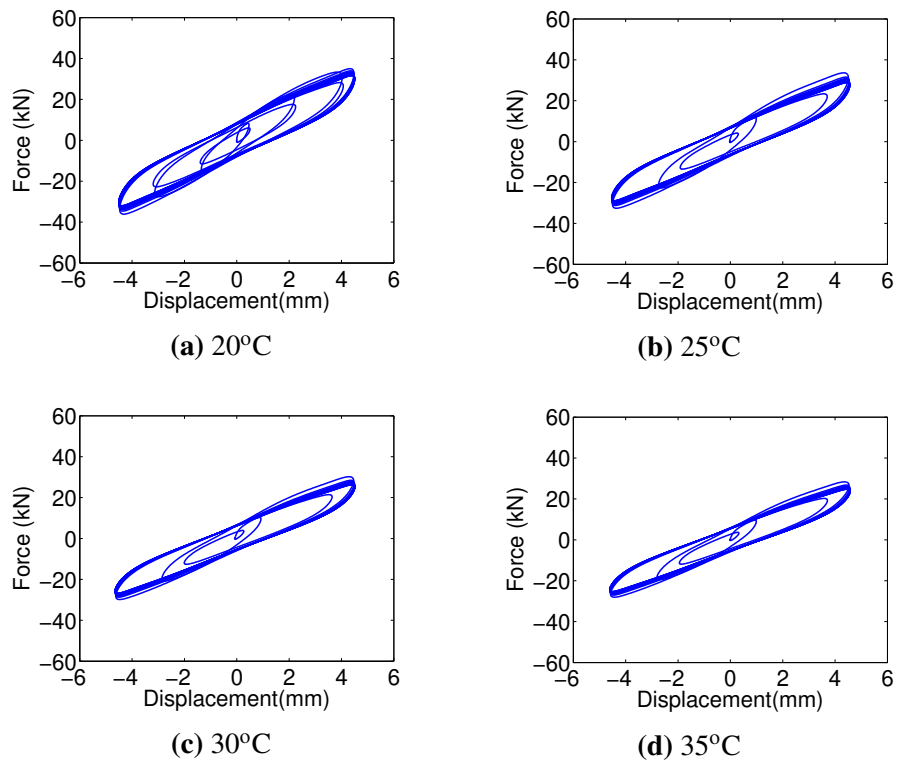


(b)

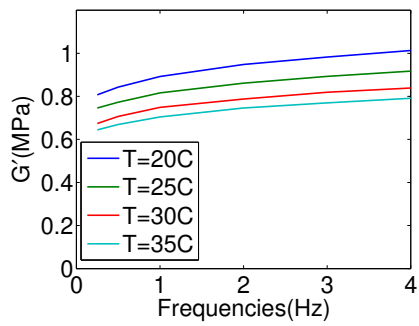
**Fig. 3.** Mechanical Properties evaluated from characterization tests under 20°C: (a) Shear storage modulus, and (b) loss factor



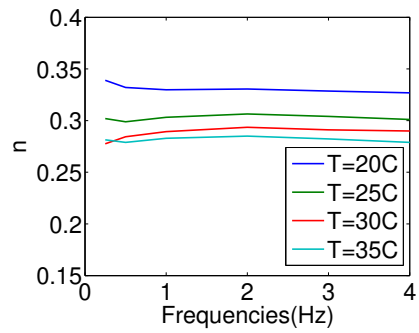
**Fig. 4.** ED's hysteresis loops for 2.0 Hz and 20°C



**Fig. 5.** Hysteresis loops for 2.0Hz, 40% shear strain for various temperatures

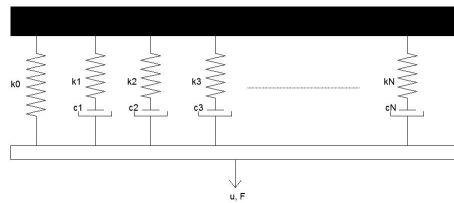


(a)

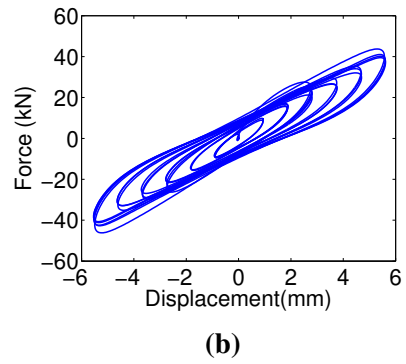
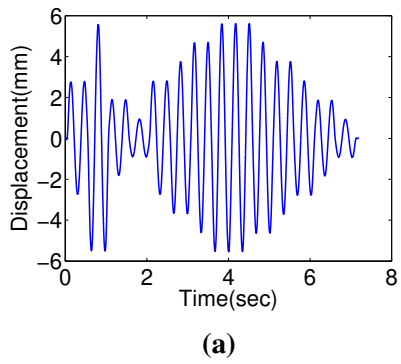


(b)

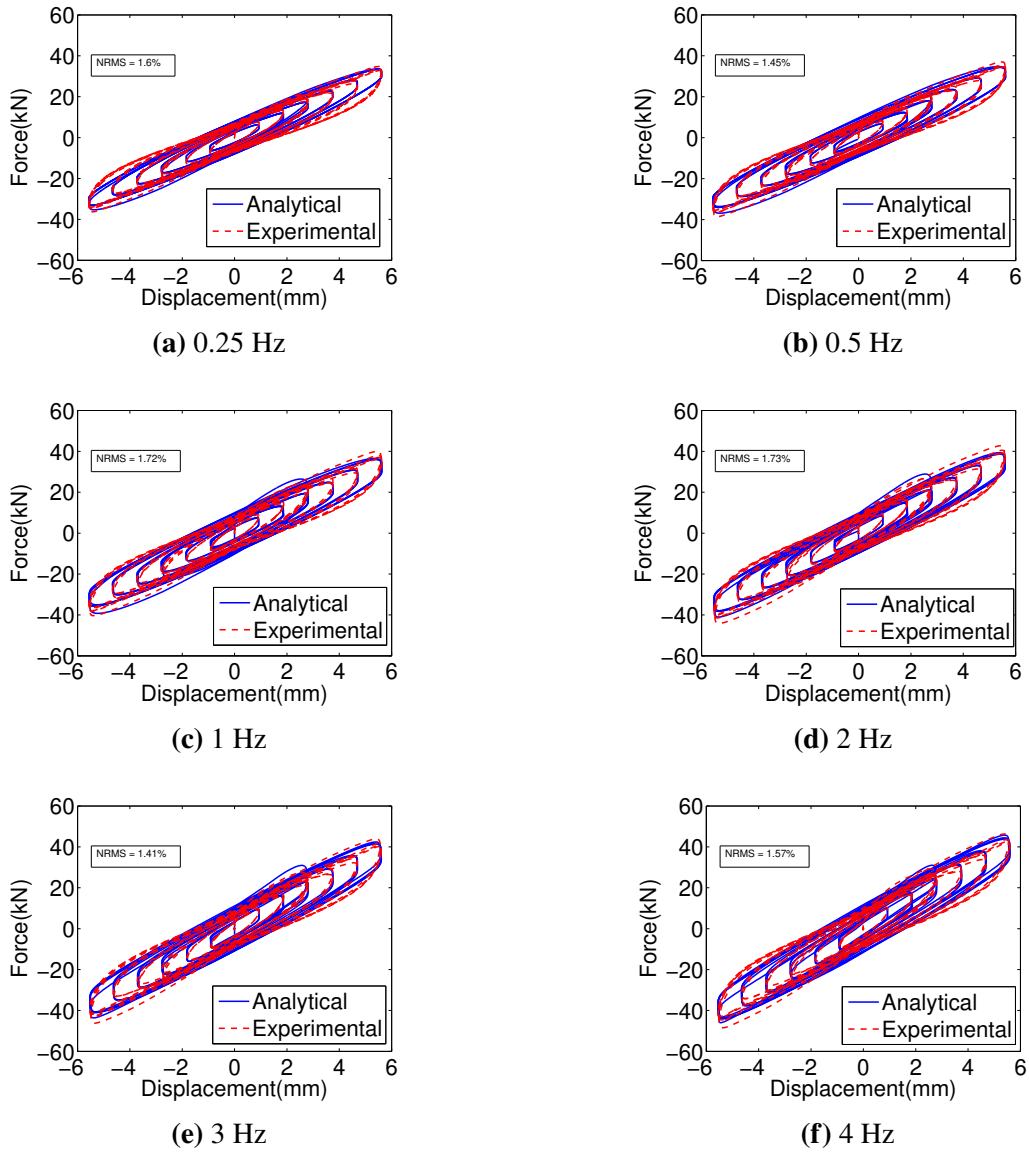
**Fig. 6.** Mechanical properties from characterization tests under 40% shear strain for various temperatures



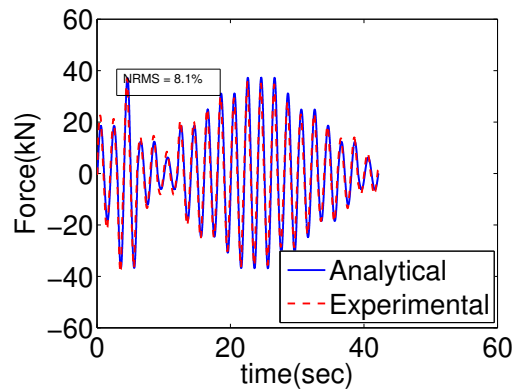
**Fig. 7.** Linear Generalized Maxwell Model (GMM)



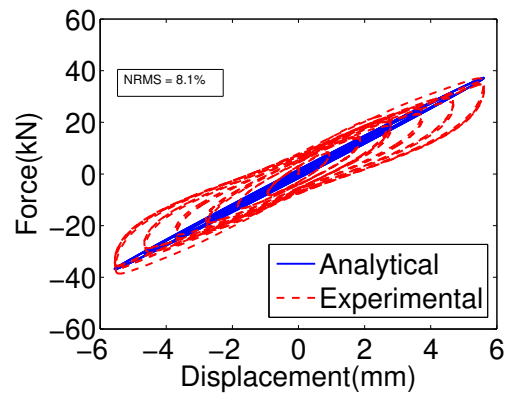
**Fig. 8.** Sweep amplitude test for frequency of 3 Hz



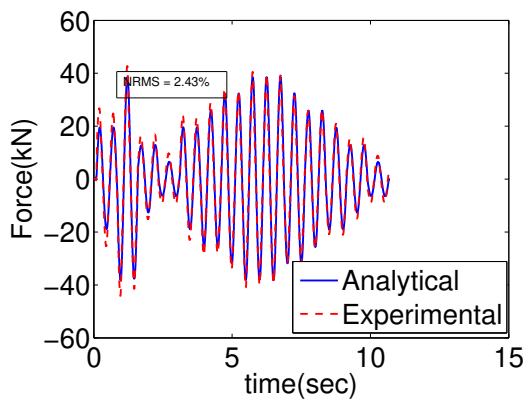
**Fig. 9.** Comparison of force between experiments and MGMM model for sweep amplitude test under 20°C



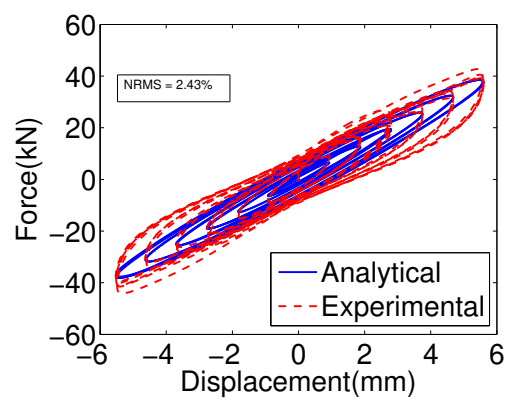
(a) Force-Time 0.5Hz



(b) Force-Displacement 0.5Hz

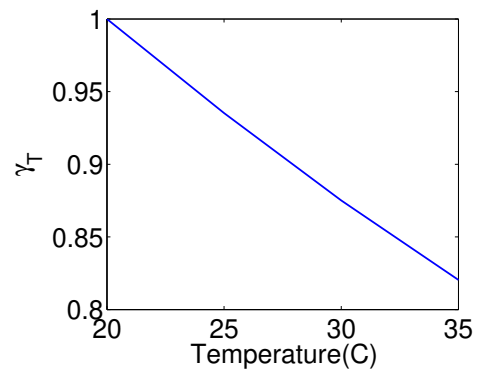


(c) Force-Time 2.0 Hz

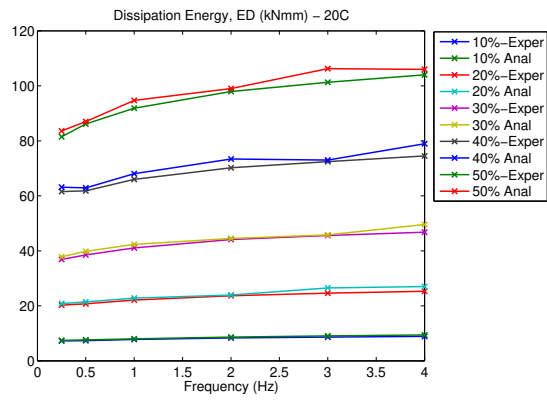


(d) Force-Displacement 2.0 Hz

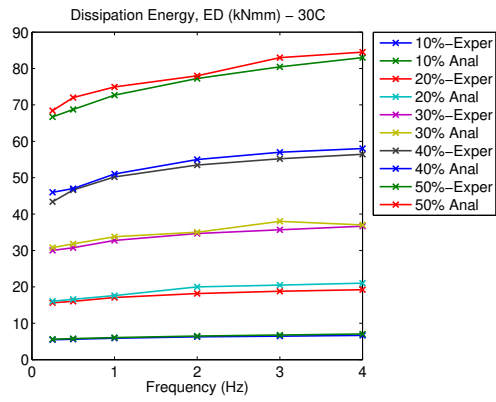
**Fig. 10.** Comparison of GMM with experimental data



**Fig. 11.** Alteration of temperature parameter  $\gamma_T$  with different temperatures



**Fig. 12.** Dissipation Energy Comparison for T=20°C



**Fig. 13.** Dissipation Energy Comparison for T=30°C